



# Gatwick Airport Northern Runway Project

Environmental Statement

Appendix 13.4.1: Air Quality Assessment Methodology

**Book 5**

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# 1 Introduction

1.1.1 This document forms **ES Appendix 13.4.1: Air Quality Assessment Methodology** of the Environmental Statement (ES) prepared on behalf of Gatwick Airport Limited (GAL) for the proposal to make best use of Gatwick Airport's existing runways and infrastructure (referred to within this report as 'the Project').

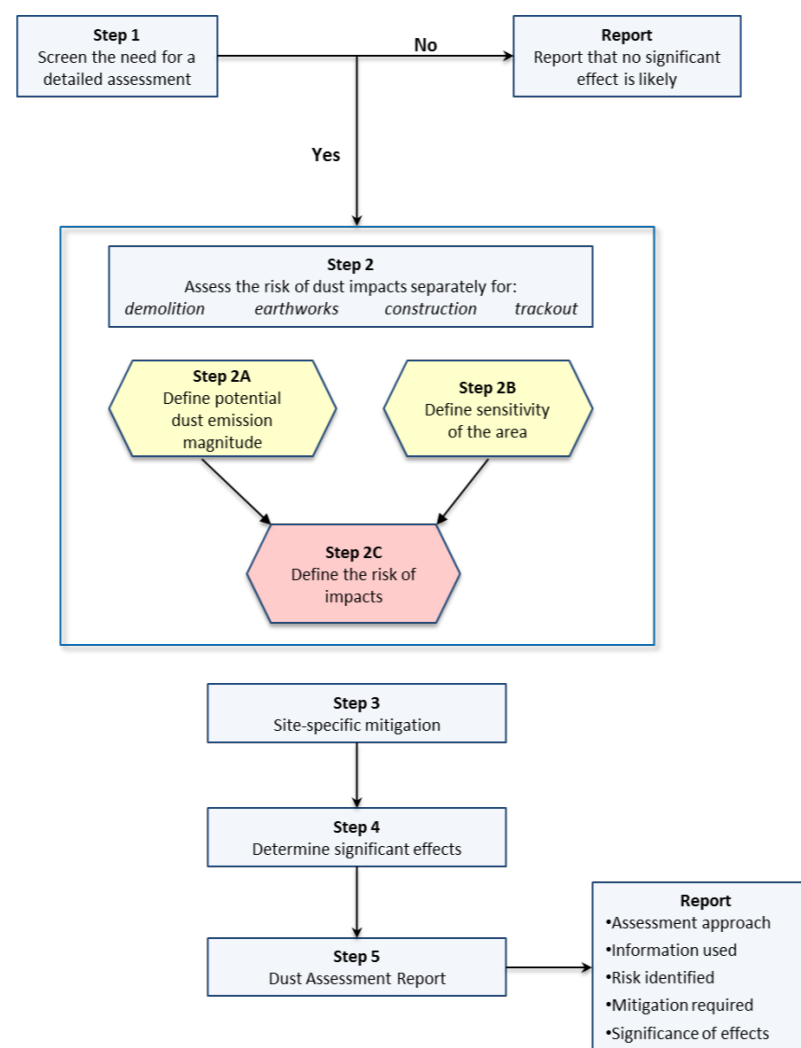
1.1.2 This document describes the methodology that has been used for the air quality and odour assessment as reported in **ES Chapter 13: Air Quality** (Doc Ref. 5.1).

# 2 Construction Dust Assessment Methodology

## 2.1 Methodology

2.1.1 There are five steps in the construction dust assessment process described in the Institute of Air Quality Assessment (IAQM) guidance (Guidance on the assessment of dust from demolition and construction) (IAQM, 2014). A description of each step is provided in this section.

**Diagram 2.1.1: IAQM Dust Assessment Methodology**



### Step 1: Need for Assessment

2.1.2 The first step is the initial screening for the need for a detailed assessment. According to the IAQM guidance (IAQM, 2014), an assessment is required where there are sensitive receptors within 350 metres of the site boundary of the scheme, or within 50 metres of ecological sites, and/or within 50 metres of the route(s) used by the construction vehicles on the public highway for up to 500 metres along the route from the site entrance(s). This approach has been followed for this assessment.

### Step 2: Assess the Risk of Dust Impacts

2.1.3 This step is split into three sections as follows:

- 2A: Define the potential dust emission magnitude;
- 2B: Define the sensitivity of the area; and
- 2C: Define the risk of impacts.

- 2.1.4 Each of the dust-generating activities is given a dust emission magnitude depending on the scale and nature of the works (step 2A) based on the criteria presented in Table 2.1.1.
- 2.1.5 The sensitivity of the surrounding area is then determined (step 2B) for each dust effect from the dust-generating activities listed in Step 2A, based on the proximity and number of receptors, their sensitivity to dust, the local particulate matter (PM<sub>10</sub>) background concentrations and any other site-specific factors. Table 2.1.2 and Table 2.1.3 show the criteria for defining the sensitivity of the area to different dust effects.
- 2.1.6 A 'high sensitivity receptor' is where "the people or property would reasonably be expected to be present continuously" such as dwellings and museums; a 'medium sensitivity receptor' is where "the people or property wouldn't reasonably be expected to be present here continuously or regularly for extended periods" such as parks and places of work; and a 'low sensitive receptor' is where "there is transient exposure, where the people or property would reasonably be expected to be present only for limited periods of time" such as footpaths and short term car parks (IAQM, 2014).
- 2.1.7 The overall risk of the impacts for each activity is then determined (step 2C) prior to the application of any mitigation measures (Table 2.1.4) and an overall risk for the site is derived.

**Table 2.1.1: Dust Emission Magnitude**

Small	Medium	Large
<b>Demolition</b>		
<ul style="list-style-type: none"> <li>Total building volume &lt;20,000 m<sup>3</sup>.</li> <li>Construction material with low potential for dust release (eg metal cladding or timber).</li> </ul>	<ul style="list-style-type: none"> <li>Total building volume 20,000 m<sup>3</sup> - 50,000 m<sup>3</sup>.</li> <li>Potentially dusty construction material.</li> <li>Demolition activities 10-20 metres above ground level.</li> </ul>	<ul style="list-style-type: none"> <li>Total building volume &gt;50,000 m<sup>3</sup>.</li> <li>Potentially dusty construction material (eg concrete).</li> <li>On-site crushing and screening, demolition activities &gt;20 metres above ground level.</li> </ul>

Small	Medium	Large
<b>Earthworks</b>		
<ul style="list-style-type: none"> <li>Total site area &lt;2,500 m<sup>2</sup>, soil type with large grain size (eg sand).</li> <li>&lt;5 heavy earth moving vehicles active at any one time.</li> <li>Formation of bunds &lt;4 metres in height.</li> <li>Total material moved &lt;20,000 tonnes.</li> <li>Earthworks during wetter months.</li> </ul>	<ul style="list-style-type: none"> <li>Total site area 2,500 m<sup>2</sup> - 10,000 m<sup>2</sup>, moderately dusty soil type (eg silt).</li> <li>5-10 heavy earth moving vehicles active at any one time.</li> <li>Formation of bunds 4 metres - 8 metres in height.</li> <li>Total material moved 20,000 - 100,000 tonnes.</li> </ul>	<ul style="list-style-type: none"> <li>Total site area &gt;10,000 m<sup>2</sup> potentially dusty soil type (eg clay, which will be prone to suspension when dry due to small particle size).</li> <li>&gt;10 heavy earth moving vehicles active at any one time.</li> <li>Formation of bunds &gt;8 metres in height.</li> <li>Total material moved &gt;100,000 tonnes.</li> </ul>
<b>Construction</b>		
<ul style="list-style-type: none"> <li>Total building volume &lt;25,000 m<sup>3</sup>.</li> <li>Construction material with low potential for dust release (eg metal cladding or timber).</li> </ul>	<ul style="list-style-type: none"> <li>Total building volume 25,000 m<sup>3</sup>-100,000 m<sup>3</sup>.</li> <li>Potentially dusty construction material (eg concrete).</li> <li>Piling.</li> <li>On-site concrete batching.</li> </ul>	<ul style="list-style-type: none"> <li>Total building volume &gt;100,000 m<sup>3</sup>.</li> <li>Piling.</li> <li>On-site concrete batching.</li> <li>Sandblasting.</li> </ul>
<b>Trackout</b>		
<ul style="list-style-type: none"> <li>&lt;10 heavy duty vehicles (HDV) (&gt;3.5 t) trips in any one day.</li> <li>Surface material with low potential for dust release.</li> </ul>	<ul style="list-style-type: none"> <li>10-50 HDV (&gt;3.5 t) trips in any one day.</li> <li>Moderately dusty surface material (eg high clay content).</li> <li>Unpaved road length 50 metres – 100 metres.</li> </ul>	<ul style="list-style-type: none"> <li>&gt;50 HDV (&gt;3.5 t) trips in any one day.</li> <li>Potentially dusty surface material (eg high clay content).</li> <li>Unpaved road length &gt;100 metres.</li> </ul>

Small	Medium	Large
<ul style="list-style-type: none"> <li>Unpaved road length &lt;50 metres.</li> </ul>		

**Table 2.1.2: Sensitivity of the Area to Dust Soiling Effects**

Receptor Sensitivity	Number of Receptors	Distance from the Source (metres)			
		<20	<50	<100	<350
High	>100	High	High	Medium	Low
	10 – 100	High	Medium	Low	Low
	<10	Medium	Low	Low	Low
Medium	>1	Medium	Low	Low	Low
Low	>1	Low	Low	Low	Low

Table 2.1.3: Sensitivity of the Area to Human Health Impacts

Receptor Sensitivity	Annual Mean PM <sub>10</sub> Concentrations	Number of Receptors	Distance from the Source (metres)					
			<20	<50	<100	<200	<350	
High	>32 µg/m <sup>3</sup>	>100	High	High	High	Medium	Low	
		10-100			Medium	Low		
		1-10			Low	Low		
	28-32 µg/m <sup>3</sup>	>100	High	High	High	Medium	Low	
		10-100			Medium	Low		
		1-10			Low	Low		
	24-28 µg/m <sup>3</sup>	>100	High	Medium	Medium	Low	Low	
		10-100			Low	Low		
		1-10			Low	Low		
	<24 µg/m <sup>3</sup>	>100	Medium	Low	Low	Low	Low	
		10-100			Low	Low		
		1-10			Low	Low		
Medium	>32 µg/m <sup>3</sup>	>10	High	Medium	Low	Low	Low	
		1-10	Medium	Low	Low	Low		
	28-32 µg/m <sup>3</sup>	>10	Medium	Low	Low	Low	Low	
		1-10	Low	Low	Low	Low		
	24-28 µg/m <sup>3</sup>	>10	Low	Low	Low	Low	Low	
		1-10	Low	Low	Low	Low		
	<24 µg/m <sup>3</sup>	>10	Low	Low	Low	Low	Low	
		1-10	Low	Low	Low	Low		
	Low	-	≥1	Low	Low	Low	Low	Low

Table 2.1.4: Risk of Dust Impacts

Sensitivity of Area	Dust Emission Magnitude		
	Large	Medium	Small
<b>Demolition</b>			
High	High Risk	Medium Risk	Medium Risk
Medium	High Risk	Medium Risk	Low Risk
Low	Medium Risk	Low Risk	Negligible
<b>Earthworks</b>			
High	High Risk	Medium Risk	Low Risk
Medium	Medium Risk	Medium Risk	Low Risk
Low	Low Risk	Low Risk	Negligible

Sensitivity of Area	Dust Emission Magnitude		
	Large	Medium	Small
<b>Construction</b>			
High	High Risk	Medium Risk	Low Risk
Medium	Medium Risk	Medium Risk	Low Risk
Low	Low Risk	Low Risk	Negligible
<b>Trackout</b>			
High	High Risk	Medium Risk	Low Risk
Medium	Medium Risk	Low Risk	Negligible
Low	Low Risk	Low Risk	Negligible

### Step 3: Determine the Site-specific Mitigation

2.1.8 Once each of the activities is assigned a risk rating, appropriate mitigation measures are identified. Where the risk is negligible, no mitigation measures beyond those required by legislation are necessary.

### Step 4: Determine any Significant Residual Effects

2.1.9 Once the risk of dust impacts has been determined and the appropriate dust mitigation measures identified, the final step is to determine whether there are any residual significant effects. The IAQM guidance notes that it is anticipated that with the implementation of effective site-specific mitigation measures, the environmental effect would not be significant in most cases.

### Step 5: Prepare a Dust Assessment Report

2.1.10 The last step of the assessment is the preparation of a Dust Assessment Report. For the ES, this is the assessment of construction dust emissions as detailed in **ES Chapter 13: Air Quality** (Doc Ref. 5.1) and **ES Appendix 13.8.1: Air Quality Construction Period Mitigation** (Doc Ref. 5.3).

## 3 Emissions Methodology

### 3.1 Overview

3.1.1 This section describes the methodology used in this assessment which builds on that used for the previous air quality assessments for Gatwick Airport in 2002/3, 2005/6, 2010 and 2015, which in turn followed the recommendations of the Department for Transport (DfT) Project for the Sustainable Development of Heathrow (PSDH) (Department for Transport, 2006). There have been updates to the methodology, specifically, accounting for reduced-engine taxiing and use of auxiliary power units (APU) off-stand for this analysis.

3.1.2 Operational air quality impacts from the airport arise as a result of emissions from aircraft traffic, other on-site activity (including the Central Area Recycling Enclosure (CARE) facility) and increased road traffic on the local road network.

3.1.3 The methodology is aimed at calculating pollutant concentrations averaged over a year for comparison with air quality standards. Concentrations over shorter averaging periods, for comparison with short-term objectives, are derived from the annual mean values using empirical relationships.

3.1.4 The airfield and road traffic contribution to air pollutant concentrations is calculated using a two-step process. The first step is the development of an emissions inventory to quantify the emissions arising from airport-related sources and road traffic, including the spatial distribution and temporal breakdown of the

emissions. Dispersion modelling is then used to calculate the contribution to ground-level concentrations at selected receptors, based on the calculated emissions, having due regard to their spatial distribution. The temporal breakdown of the emissions ensures that meteorological conditions are applied properly.

3.1.5 The aim of the inventory methodology is to generate a realistic best estimate of the emissions. Where possible, activity data for the calendar year 2018 were used to align with the base year for model verification. Where such activity data were not available, statistics for the nearest available period were used, and adjusted as necessary.

### Pollutants Assessed

3.1.6 In common with most activities involving the combustion of fuel, activities associated with an airport release a wide variety of pollutants but, for most of the regulated pollutants, airport emissions (even from a large airport) do not have the potential to be a significant factor in whether or not current air quality standards can be met around the airport. The relevant evidence was previously reviewed by the air quality technical panels set for the PSDH (Department for Transport, 2006). Based on the available monitoring and modelling data, it was concluded that benzene, 1,3-butadiene, carbon monoxide, lead, polycyclic aromatic hydrocarbons (PAHs) and sulphur dioxide (SO<sub>2</sub>) were not priority pollutants at airports, leading to a focus on oxides of nitrogen (NO<sub>x</sub>), particulate matter (PM<sub>10</sub> and PM<sub>2.5</sub>) and ozone (O<sub>3</sub>). O<sub>3</sub> is not a primary airport pollutant, although airports contribute precursors (volatile organic compounds (VOCs) and nitrogen dioxide (NO<sub>2</sub>)) to the formation of O<sub>3</sub> on a regional and

trans-national scale. Therefore, O<sub>3</sub> is not currently included in the regulations for local air quality management (The Air Quality Standards Regulations, 2016) and is not considered in this assessment. Although the PSDH (Department for Transport, 2006) review of priority pollutants was carried out in the Heathrow Airport context, the reasoning is also transferable to Gatwick and has been applied in air quality assessments of other major airports in the United Kingdom (UK).

3.1.7 The main pollutants included in this assessment are therefore NO<sub>x</sub>, NO<sub>2</sub>, PM<sub>10</sub> and PM<sub>2.5</sub>. As a result of the operation of the CARE facility, the associated pollutants assessed are NO<sub>2</sub>, PM<sub>10</sub>, PM<sub>2.5</sub>, carbon monoxide (CO), SO<sub>2</sub> and VOCs. Also, due to the combined effect to nitrogen deposition at sensitive ecological sites ammonia (NH<sub>3</sub>) has been assessed from road traffic.

3.1.8 The NO<sub>x</sub> emitted from combustion sources is principally in the form of nitric oxide (NO), with usually only a small percentage of NO<sub>2</sub> directly emitted from the combustion source (ie primary NO<sub>2</sub>) (pNO<sub>2</sub>). After release, further NO<sub>2</sub> is formed in the atmosphere by transformation of NO, principally as a result of the reaction with ambient O<sub>3</sub>; the fraction of NO converted to NO<sub>2</sub> at various distances from the source depends on a number of climatological factors. pNO<sub>2</sub> fractions for the aircraft sources were taken from the methodology of the PSDH (Department for Transport, 2006) and are shown in Table 3.1.1.

**Table 3.1.1: pNO<sub>2</sub> Fractions for Aircraft Emissions**

Thrust Setting	pNO <sub>2</sub> Fraction
100%	4.5%
85%	5.3%
30%	15.0%
7%	37.5%

3.1.9 In relation to PM<sub>2.5</sub> emissions, the European Monitoring and Evaluation programme (EMEP)/European Environment Agency (EEA) Guidebook (EMEP/EEA, 2019) states that “it is reasonable to assume that for aircraft, their PM emissions can be considered as PM<sub>2.5</sub>”. Therefore, it was assumed that all particulate matter emissions from aircraft engines were in the PM<sub>2.5</sub> fraction. For the road sources, emission factors for PM<sub>2.5</sub> are available so no assumption about the PM<sub>2.5</sub> fraction from road traffic was required.

3.1.10 Where different assumptions on the calculation of pNO<sub>2</sub> and PM<sub>2.5</sub> emissions from those given in paragraphs 3.1.8 and 3.1.9 have been made for other emissions sources, these are reported in the following sections.

### 3.2 Sources of Emissions

3.2.1 An inventory of NO<sub>x</sub>, pNO<sub>2</sub>, PM<sub>10</sub> and PM<sub>2.5</sub> emissions was built for the following pollution sources:

- aircraft main engines in the landing and take-off (LTO) cycle on the ground and up to a height of 3,000 ft (915 metres);
- aircraft auxiliary power units (APU);
- aircraft engine testing;
- ground support equipment (GSE);
- airport heating plant;
- fire training ground (FTG);
- road vehicles on the local and strategic highway network around the airport and at car parks; and
- CARE facility

3.2.2 For PM<sub>10</sub> and PM<sub>2.5</sub>, the inventory includes not only exhaust emissions but also fugitive emissions from brake and tyre wear for aircraft and road traffic. Also NH<sub>3</sub> was derived from road traffic emissions.

### 3.3 Aircraft Emissions During the LTO Cycle

3.3.1 The dominant aircraft source of emissions is main-engine exhaust during the LTO flight phases (modes). Separate consideration is given to emissions from APUs and engine testing (engine ground runs).

3.3.2 The contribution to aircraft exhaust emissions (in kg) arising from a given mode of aircraft operation from a single engine is given by the product of the duration (in seconds) of the operation, the engine fuel flow rate at the appropriate thrust setting (kg fuel per second) and the emission factor for the pollutant of interest (kg pollutant per kg fuel). The annual emissions total for the mode (kg per year or tonnes per year) is obtained by summing contributions over all engines for all aircraft movements in the year.

### LTO Flight Phases

3.3.3 The following ‘modes’ (phases) of the LTO cycle are considered for the purpose of emissions estimation:

- approach (from 3,000 ft altitude to runway threshold);
- landing roll (from runway threshold to runway exit);
- taxi-in;
- taxi-out;
- hold at runway head;
- take-off roll (from start-of-roll to wheels-off);
- initial climb (from wheels-off to throttle-back); and
- climb-out (from throttle-back to 3,000 ft altitude).

3.3.4 ‘Taxi-out’ commences at stand (including pushback) and ends when the aircraft joins the departure queue; ‘taxi-in’ commences when the aircraft leaves the runway and ends when the aircraft reaches the stand. There may be some overestimation of taxi-out emissions from assuming all engines are lit during pushback, but there is a lack of information on when engines are lit as a function of aircraft type and operator. It is assumed that all engines are shut down immediately when the aircraft reaches the stand<sup>1</sup>. It is judged that, on average, any potential underestimation of aircraft emissions from this assumption is compensated by the assumption that all engines are lit during pushback.

3.3.5 Helicopters do not have take-off roll or landing roll, and a single mode covers the climb from ground to an altitude of 3,000 ft.

### Reduced-engine Taxiing

3.3.6 Reduced-engine taxiing is the practice of shutting down an engine during taxiing operations, which helps reduce fuel use, emissions, and noise. In theory, reductions of 20 to 40 % of the ground level fuel burn and carbon dioxide (CO<sub>2</sub>), and 10 to 30 % of ground level NO<sub>x</sub> emissions, may be realised depending on aircraft type and operator technique. However, some of the reductions may be offset by the need to keep the APU running during taxiing.

3.3.7 For this assessment, a survey of the airlines was undertaken to identify the extent to which reduced-engine taxiing was used at the airport. Responses to the survey showed the practice of reduced-engine taxiing to be common at Gatwick and provided estimates of the frequency and duration for both arrivals and departures on an airline/aircraft fleet by fleet basis. These data

<sup>1</sup> It is recognised that some engines may have been shut down prior to arrival at stand if the aircraft is operating reduced-engine taxiing.

have been included in the emission calculations, with suitably averaged data applied to those airline/aircraft fleets for which survey information was not available.

### Movement Data and Fleet Mix

3.3.8 Detailed information on flight-by-flight records for the baseline year of 2018 was provided by GAL from its aircraft movement database. This included:

- actual flight date and time;
- arrival or departure identifier;
- aircraft type;
- stand number;
- runway number;
- aircraft registration number;
- operator; and
- aircraft engine (in the form of a unique engine identifier (UID)).

### Engine Assignment

3.3.9 GAL's aircraft movement database includes a UID, which, for jet aircraft, links directly to records in the International Civil Aviation Organisation (ICAO) databank of emission factors (European Union (EU) Aviation Safety Agency (EASA), 2021). For a small fraction of movements, the UID was unknown or erroneous; for these, a default engine was assigned based (where possible) on the most common engine for that aircraft type at Gatwick Airport. Where there was no instance in the Gatwick data giving an engine assignment for a particular aircraft type, a typical engine was chosen according to standard aircraft reference sources.

### Exhaust Emission Factors

3.3.10 The emission factors for aircraft engines vary from one engine type to another, and, for a given engine, depend on thrust setting. The main source of emission factors (and fuel flow rates) used in the assessment is the ICAO databank, which gives certification test results for most of the jet engines in service at four thrust settings (7 %, 30 %, 85 % and 100 % of rated thrust) (EASA, 2021). Data for some engines not listed in the ICAO databank were obtained from the FOI (Swedish Defence Research Agency) compilation (FOI, 2002) for turboprops or Federal Office of Civil Aviation (FOCA) piston engine database and helicopter emissions table.

3.3.11 Certification data in the ICAO databank are based on tests carried out using new or nearly-new production engines, with

certification data corrected to production standard (EASA, 2021). Thus, the applicability of certification data to in-service engines requires consideration. For reasons of safety and fuel efficiency, aircraft engines operate within closely monitored ranges of tolerance and are subject to strict maintenance schedules. In the past, uncertainties in emission rates related to engine ageing were judged as small compared to other uncertainties and were not taken into account. However, at any particular time the engines in the fleet operating at an airport would be, on average, part-way through the maintenance cycle; in addition, there would be some longer-term degradation not restored by maintenance that would be restored only at refurbishment. Thus, there may be a systematic bias in emissions estimates based on certification data.

3.3.12 The available data on this issue were reviewed by QinetiQ for the PSDH (Department for Transport, 2006), in particular distinguishing whole-flight deterioration values from LTO-only values, leading to a recommendation of a 4.3 % increase in fuel flow rates in the LTO cycle compared to certification values and a 4.5 % increase in NO<sub>x</sub> emission rates (the product of fuel flow rate and emission index) compared to certification values. Although there was some indication in the available data of variation with engine type, the data were not detailed enough to support engine-specific recommendations: the values given are appropriate averages for the fleet as a whole, bearing in mind the range of engine age in the fleet at any given time. These fleet-averaged values, applied to Heathrow in the PSDH work, were judged equally applicable to Gatwick.

3.3.13 The available data are also not detailed enough to make a distinction between the various phases of the LTO cycle (taxiing, take-off) so, in applying these values in the PSDH work, the percentage NO<sub>x</sub> increase was applied equally to the NO<sub>x</sub> emissions from all phases. It was recommended that the fuel increase be applied to PM<sub>10</sub> emission rates, recognising the major uncertainties in PM<sub>10</sub> emission indices (further detailed below). These recommendations were applied to this assessment.

3.3.14 The ICAO databank contains measured non-volatile PM<sub>10</sub> emission factors for only a small number of newer engines. For older engines it only includes 'smoke number' (SN). This is an indirect measure of particulate emissions calculated from the reflectance of a filter paper measured before and after the passage of a known quantity of smoke-bearing gas. For the PSDH, methods and data for deriving aircraft exhaust PM<sub>10</sub> emission indices were reviewed by QinetiQ, and

recommendations were made for an interim methodology to be used while further data were being collected from various programmes in several countries. A closely similar methodology has been advocated in guidance by the ICAO Committee on Aviation Environmental Protection (CAEP) on the calculation of airport emission inventories (CAEP, 2004). This includes a means of deriving non-volatile PM<sub>10</sub> emission factors from SN, which has been adopted for older engines for this assessment, and methodologies for estimating the volatile sulphate and organic PM<sub>10</sub> component, which have also been adopted for this assessment.

3.3.15 The ICAO certification test results are given at the four standard thrust settings (7 %, 30 %, 85 % and 100 % of engine rating), whereas recent airport inventories take account of differences between actual thrust settings and the ICAO set points, particularly for take-off thrust. The ICAO CAEP committee has issued a guidance note on the use of the ICAO databank in assessing airport emissions, which included advice on calculating emission indices at intermediate thrust settings (CAEP, 2004). If the fuel flow rate at the intermediate setting is known, the preferred method of interpolation is the 'Boeing fuel flow method' (Baughcum *et al.*, 1996), which interpolates the emission index as a function of the fuel flow rate; however, actual take-off fuel flow rates are not generally available for Gatwick operations. In this case, CAEP gives guidance on how to interpolate emission index on the basis of thrust value, suggesting a multi-order polynomial for NO<sub>x</sub> (but also noting that linear interpolation between 100 % and 85 % thrust has good accuracy in this range). The PSDH report (Department for Transport, 2006) endorsed the multi-order polynomial approach for NO<sub>x</sub> in the absence of actual fuel flow rate data, and this approach was adopted for this assessment. The fuel flow rate and SN have therefore been calculated using interpolation.

### Effect of Ambient Conditions

3.3.16 Aircraft engine emissions (NO<sub>x</sub> in particular) vary with ambient temperature, pressure and humidity. The certification test results in the ICAO databank are corrected to sea-level international standard atmosphere conditions (EASA, 2021). The CAEP guidance note considered the effect of variations in ambient conditions, noting that variations in ambient pressure and temperature are primarily reflected in changes in operating conditions and are therefore largely taken into account when actual thrust settings are used instead of notional ones; thus, no additional adjustment was recommended.



- 3.3.17 However, there would be some variation in NO<sub>x</sub> emission rates (ie the product of fuel flow rate and emission index) with hour-to-hour variations in ambient conditions because of the associated changes in engine operating point. This was examined by QinetiQ as part of the PSDH work, leading to a technical report (Horton, 2006) which recommends a method for adjusting NO<sub>x</sub> emission rates at a given thrust to ambient temperature and pressure. The sensitivity to ambient temperature and pressure variations was found to be significantly greater for the higher overall pressure ratio (OPR) engines (40:1 and above) that are now common on modern large jets (for example, the Rolls-Royce Trent 1000 engine as fitted to the Boeing 787 aircraft has OPR values of up to 49.4). QinetiQ estimated that the impact on total ground-level NO<sub>x</sub> emissions over the year, using weather data for Heathrow in 2002, is typically in the order of a few %. However, annual-average emission rate is not the only parameter of interest in air quality assessment, even when calculating annual-mean concentrations: the diurnal and seasonal variation in emissions is also important, given that the frequency of meteorological conditions leading to better (or worse) atmospheric dispersion varies with hour of day and month of year. QinetiQ found that, for the most sensitive type of engine, the hourly NO<sub>x</sub> emission rate at a given thrust varied during a year by up to ±50 % from the value calculated assuming International Standard Atmosphere (ISA) conditions.
- 3.3.18 The QinetiQ report found that it was not possible to condense the results of their analysis into simple expressions applicable to a small number of engine type categories because of wide variations from one individual engine to another (Horton, 2006). Therefore, a calculation method that derives factors to apply to emissions of NO<sub>x</sub>, hydrocarbons (HC) and CO for each engine type in the ICAO databank was implemented, covering a wide range of ambient pressures and temperatures (EASA, 2021). For the remainder of engines (principally turboprops) QinetiQ default parameters were used.
- 3.3.19 In light of the relatively poor characterisation of aircraft PM emissions, the PSDH report recommended that no adjustment for variations in ambient conditions be applied to PM emission rates (Department for Transport, 2006).
- 3.3.20 The temperature and pressure variation with altitude would affect emission rates during climb and approach for an individual flight. As the aircraft climbs or descends, there are continuous changes in forward speed, temperature and pressure to which the engine control system would respond appropriately. However, emissions at increasing height have a decreasing impact on ground-level concentrations, which are the principal focus of interest in local air quality assessment. Even bearing in mind the potential impact of trailing vortices in transporting exhaust gases downwards, it is unlikely that emissions above 200 metres height have a significant impact on ground-level concentrations. For this reason, greater effort has been put into realistically representing the emission rates for the lowest few hundred metres in height than for greater heights.
- 3.3.21 To address this, the NO<sub>x</sub> emission rate during the initial-climb phase of the LTO cycle (from wheels-off to engine cut-back, typically at 1,000 ft to 1,500 ft) was calculated based on the ground-level temperature and pressure. This ensures that the emission rate in the lowest part of the initial climb is not underestimated, accepting that there would be some slight overestimation of the average emission in the initial climb taken over the whole year. For the climb-out phase (from cut-back height to 3,000 ft), the hourly surface temperature and pressure values were adjusted using simple representative profiles of temperature and pressure. Temperature was assumed to decrease with height from its surface value in line with the dry adiabatic lapse rate of -9.8°C per km (which would only strictly be the case for zero heat flux to/from the ground); the temperature adjustment to climb-out emissions was worked out using the mid-height temperature for the climb-out phase. Pressure was assumed to vary with height in a manner consistent with the adiabatic lapse rate for an atmosphere in hydrostatic equilibrium. This simpler procedure for climb-out emissions is judged adequate for emissions in this part of the LTO cycle, which have an insignificant impact on ground-level concentrations.
- 3.3.22 Similar simple procedures were used to account for the temperature/pressure variation with altitude during approach.
- 3.3.23 For correcting from NO<sub>x</sub> test results in the databank to actual humidity, the CAEP document advocates using in reverse the expression provided by ICAO Annex 16 Vol II (ICAO, 1993) to adjust test results to ISA conditions, albeit correcting a slight error in the reference specific humidity quoted in Annex 16 (ICAO, 1993). This adjustment is engine independent. Typically, this leads to hourly variations in the ground-level NO<sub>x</sub> emission rate over the year for a given thrust setting of around ±5 %, although the net effect on total annual emissions is much less. The adjustment for relative humidity is given by:
- $$EI(NO_x)_{adjusted} = EI(NO_x)_{ICAO} \exp(19(H_{ref} - H))$$
- 3.3.24 For elevated emissions, it was assumed that the specific humidity is constant with height, which is strictly true only in the absence of condensation and evaporation.
- 3.3.25 The hourly surface temperature and humidity data was taken from meteorology data for 2018 at Gatwick Airport. Atmospheric surface pressure data, which is not included in this dataset, was obtained from the National Oceanic and Atmospheric Administration website (NOAA, 2019).
- Forward-Speed Effect**
- 3.3.26 Emission indices and fuel flow rates in the ICAO databank are measured on a stationary engine in a test cell. Generally, there would be a difference in the emission rate (the product of fuel flow rate and emission index) at a selected take-off thrust when the aircraft is moving at speed with respect to the air drawn into the engine compared to the emission rate for an aircraft that is stationary.
- 3.3.27 To estimate the effect of forward speed on NO<sub>x</sub> emission rate, the approach specified by QinetiQ was similar to that for estimating the effect of ambient temperature and pressure variations, with the key influence being the effect of forward velocity on the relative temperature and pressure at the engine inlet. The results of the analysis are given in the QinetiQ report (Horton, 2006). The principal effect of interest from a local air quality viewpoint is the change in emission rate during the take-off roll, although consideration was also given to the effect of forward speed on climb and approach emissions. The aircraft engine management system would respond to the inlet changes experienced. For example, QinetiQ assumed a representative 1.1 % increase in fuel flow over the roll, based on samples of Flight Data Recorder (FDR) data. Thus, the forward-speed adjustment to emission rates is the combined effect of changes in fuel flow rate and changes in emission indices.
- 3.3.28 The net impact of these changes is that the NO<sub>x</sub> emission rate increases with increasing speed during the take-off roll, with the fractional increase tending to be greater for engines with higher OPR. Table 3.3.1 presents the calculated ratio of emission rate at the end of the roll to the static emission rate at full thrust for a sample of common engine types. For engines with OPR around 40 the factor at the end of roll is around 1.15 (ie a 15 % higher emission rate).

**Table 3.3.1: Mean and Final NO<sub>x</sub> Factors During Take-off, for a Range of Engine Types**

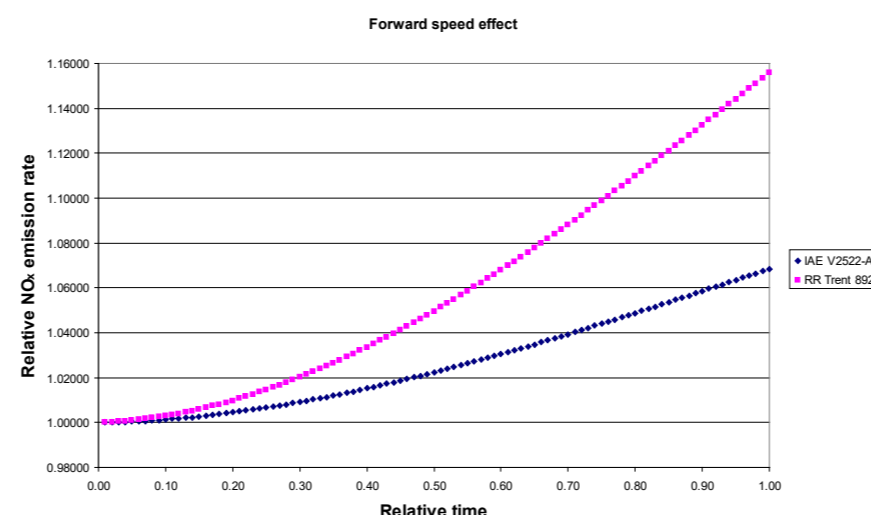
Engine	OPR <sup>1</sup>	Mean Factor <sup>2</sup>	Final Factor <sup>2</sup>
CFM56-3C-1	25.5	1.0251	1.0645
V2527-A5	27.2	1.0272	1.0700
CFM56-5B3/P	32.8	1.0367	1.0950
Trent 772	35.8	1.0505	1.1314
Trent 892	41.4	1.0590	1.1542

<sup>1</sup> OPR – overall pressure ratio

<sup>2</sup> ‘Factor’ is the ratio of NO<sub>x</sub> emission rate accounting for aircraft speed to that for stationary aircraft

3.3.29 The above approach was implemented for the impacts of forward speed on engine emissions in the same calculation tool as used for the ambient condition effects. For each engine type, the factors on emissions are calculated as coefficients of a cubic polynomial representing the emission rate as a function of time, with the emission rate expressed relative to the static emission rate at the selected take-off thrust and time expressed as a fraction of total roll time. In principle, this normalised emission profile depends on the actual take-off thrust selected, but the PSDH report found that the relevant factors for 85 % thrust were close to those for 100 % thrust. Thus, a single normalised profile is assumed to apply for a given engine to all take-off thrust values. For illustration, Diagram 3.3.1: Example of Forward Speed Effect for NO<sub>x</sub> Emissions During the Take-off Roll. NO<sub>x</sub> Emission Rate is Relative to the Value for a Stationary Aircraft; Time is Expressed as a Fraction of the Total Roll Time presents the profile for two common engines of widely different OPR.

**Diagram 3.3.1: Example of Forward Speed Effect for NO<sub>x</sub> Emissions During the Take-off Roll. NO<sub>x</sub> Emission Rate is Relative to the Value for a Stationary Aircraft; Time is Expressed as a Fraction of the Total Roll Time<sup>2</sup>**



3.3.30 Forward-speed effects are also considered during the initial climb, climb out and approach phases of the LTO cycle. For the initial climb phase, the forward-speed factor worked out for the end of the take-off roll was applied. The tool used for the calculation of the factors applied during take-off also derived those for the climb-out and approach phases, calculated using a representative speed and thrust level for each phase. Thus, the forward-speed adjustments for these phases are treated more approximately than for the take-off roll, with the same justification as that given in paragraph 3.3.20 in the context of adjustment for ambient conditions.

3.3.31 There was insufficient information available in the PSDH to quantify the effect of forward speed on PM<sub>10</sub> emission rates and it recommended that the effect is disregarded for this pollutant; correspondingly, the impact on PM<sub>2.5</sub> emissions was also disregarded.

### Engine Spool-Up

3.3.32 In the compilation of emission inventories prior to the PSDH work, it was assumed that the selected take-off thrust is applied immediately at the start of take-off roll. In practice, there is a period of engine ‘spool-up’ during which fuel flow rates and thrust levels are significantly less than the take-off values. The duration

of this initial phase depends on aircraft type, and for large aircraft may be in the order of 10 seconds, which is a significant portion of the total roll time (around 40 seconds).

3.3.33 Although the engine thrust is significantly less than take-off thrust during this phase, the engine is not at equilibrium, and it is difficult to predict what the effective emission index (kg pollutant per kg fuel burned) would be, even if the fuel flow rate is known. Thus, the PSDH made an interim recommendation that the NO<sub>x</sub> emission index be held the same during the transient phase as that applicable at take-off thrust, so the net effect of spool-up on estimated emission rate derives solely from the lower fuel flow rate.

3.3.34 QinetiQ examined FDR data obtained during take-off for a number of aircraft types and found that the data on fuel flow rate versus time collapsed reasonably well onto a single curve when fuel flow rate was expressed as a fraction of the flow rate at take-off thrust and time was expressed as a fraction of total roll time (Horton, 2006). For ease of implementation, this curve was fitted using a simple analytic expression of the form:

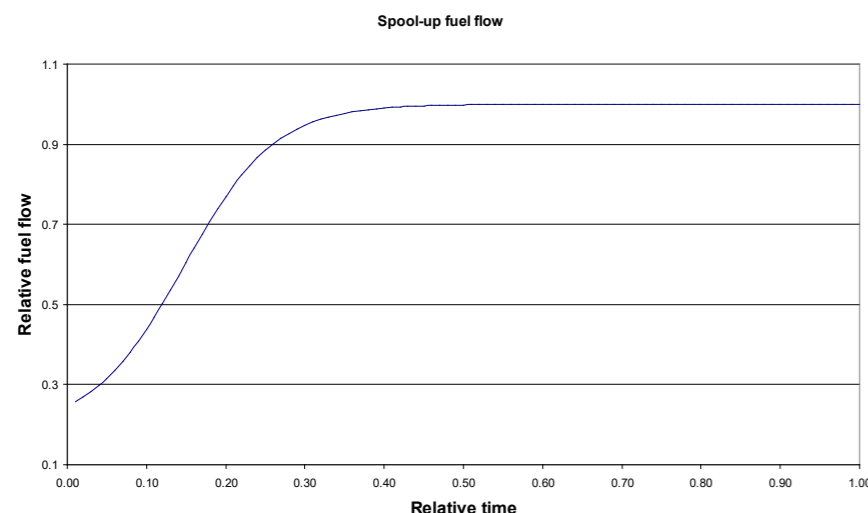
$$f(t) = a \tanh(bt + c) + d$$

where  $f(t)$  is the fuel flow rate expressed as a fraction of flow rate at take-off thrust and  $t$  is time expressed as a fraction of total roll time.  $\tanh$  denotes the hyperbolic tangent function;  $a$ ,  $b$ ,  $c$  and  $d$  are constant, with the values  $a = 0.405$ ;  $b = 8.720$ ;  $c = -1.282$ ;  $d = 0.595$ . This form, which is shown in Diagram 3.3.2: Fuel Flow Variation due to Engine Spool-up During Take-off Roll was adopted by the PSDH and has been applied to all engines and aircraft types in compiling the 2018 Gatwick Airport inventory of NO<sub>x</sub> emissions.

<sup>2</sup> The relative emission rates shown in the Diagram 3.3.1. NO<sub>x</sub> Emission Rate is Relative to the Value for a Stationary Aircraft; Time is Expressed as a Fraction of the Total Roll Time account

solely for the effect of forward speed and do not include the effect of engine spool-up (see later). In implementation, both effects are taken into account.

**Diagram 3.3.2: Fuel Flow Variation due to Engine Spool-up During Take-off Roll<sup>3</sup>**



3.3.35 For PM<sub>10</sub>, there are even greater uncertainties in SN during the transient spool-up phase than in the NO<sub>x</sub> emission index. Given the overall uncertainties surrounding the calculation of PM<sub>10</sub> emission rates, the PSDH recommended that the effect of spool-up be ignored for this pollutant, ie take-off thrust is assumed to apply from the start of roll. This recommendation has been followed in this assessment and has also been applied to PM<sub>2.5</sub> emissions.

### Thrust Settings

#### Approach

3.3.36 In the standard ICAO LTO cycle, approach thrust is set at 30 % throughout the descent from 3,000 ft to touchdown, as shown in Table 3.3.1. Although some FDR data analysed in the EU Aircraft Environmental Impacts and Certification Criteria (AEROCERT) programme (Middel, 2001) indicated that in practice thrust levels were often less than 25 % and were variable during the approach, it was considered adequate from a local air quality perspective to retain the 30 % value in airport emission inventories, given that most of the approach emissions are well above the ground.

3.3.37 However, in line with its intention of improving estimates of elevated LTO emissions as well as near-ground emissions, the PSDH defined a typical approach procedure at Heathrow as follows. Aircraft follow a 3° glide path (as in previous

assessments) with power levels of 15 % of maximum thrust from 3,000 ft down to 2,000 ft and 30 % of maximum thrust from 2,000 ft to touchdown. This requires the approach to be treated in two sections with differing emission rates. Although devised for Heathrow, it was judged that this generic approach prescription is adequately representative of Gatwick operations and has therefore been applied in this assessment.

### Reverse Thrust on Landing

3.3.38 Some arriving aircraft deploy thrust reversers at thrust levels above idle on landing whereas other aircraft, although they may deploy the reversers, use only idle thrust and rely on the wheel brakes to slow down the aircraft. There are three key parameters determining the total annual emissions from landing roll: the fraction of aircraft of a given type that use reverse thrust on landing; the duration of reverse-thrust deployment; and the thrust level engaged.

3.3.39 For this Project, GAL undertook a survey of the airlines to identify the extent to which reverse thrust was used on landing at Gatwick. Responses to the survey provided estimates of the frequency and duration of reverse idle and reverse thrust above idle on an airline/aircraft fleet by fleet basis. These data have been included in the emission calculations with suitable averaged data applied to those airline/aircraft fleets for which data were not available.

### Taxiing

3.3.40 Taxiing is assigned a thrust setting of 7% in the standard ICAO LTO cycle. However, there is evidence that actual taxiing thrust settings are on average less than this. However, it is unclear how emission indices would behave at lower thrust settings. For the products of incomplete combustion, such as CO and HC, the emission indices (g pollutant per kg fuel burned) are likely to be higher for lower thrust settings, with the reverse likely to be true for NO<sub>x</sub>; the position for SN and PM<sub>10</sub> emission indices is unclear. Lower taxiing thrust was partly taken into account in the 2002/3 Gatwick Airport emission inventory in that taxiing fuel flow rates were provided by British Airways (BA) for all the major aircraft types in their fleet, derived from information in their fuel management databases. These data confirmed that aircraft were on average taxiing at less than seven % thrust. However, it was not clear if the BA dataset could be extended to other airlines, so it was applied only to BA movements. Emission indices (g per kg)

were held at the values for seven % thrust, recognising that this might lead to overestimation of NO<sub>x</sub> emissions.

3.3.41 The evaluation of taxiing emissions is made potentially more complex by the practice of reduced-engine taxiing, which is favoured by some operators for some aircraft types. For this Project, GAL undertook a survey of the airlines to identify the extent to which reduced-engine taxiing was used at the airport. Responses to the survey showed the practice of reduced-engine taxiing to be common at Gatwick and provided estimates of its frequency and duration for both arrivals and departures on an airline/aircraft, fleet by fleet basis.

3.3.42 For taxiing on all engines, the PSDH recommended that idle thrust settings lower than seven % should be taken into account. FDR data compiled for the PSDH indicated that in most cases the ground-idle thrust setting used during most of taxiing and hold was around five %, except for aircraft fitted with Rolls-Royce engines, for which three % thrust was a closer approximation. Clearly, there would be brief periods of higher thrust (perhaps 10 to 15 %) to get the aircraft rolling or to negotiate sharp turns, but they are superimposed on much longer periods at the ground idle setting, so the average thrust level would be significantly below seven %.

3.3.43 It is easier to estimate the impact of these lower thrust settings on fuel flow than on emission indices. Considering the available data as a whole, the PSDH recommended that fuel flow rates for engine types other than Rolls Royce be set 15 – 20 % lower than the ICAO seven % value and for Rolls Royce engines be set 30 – 35 % lower than the ICAO seven % value, and these recommendations were implemented for Heathrow by using the mid-point of the ranges, ie 17.5 % and 32.5 % respectively, with the values applied to all periods of taxiing and hold. The PSDH further recommended that the NO<sub>x</sub> and PM<sub>10</sub> emission indices at the lower fuel flow rate be held the same as the value at seven % thrust. As noted earlier, this is likely to yield a somewhat conservative estimate (ie overestimate) of taxiing NO<sub>x</sub> emissions; current information (QinetiQ, 2006), albeit more uncertain, suggests that this assumption is also likely to be conservative for PM<sub>10</sub>. These recommendations were adopted in this assessment.

3.3.44 Analysis of the impact of reduced-engine taxiing on emissions suggests that the engines that are in use generally have to be operated at higher thrust settings (and the APU may be running for longer). In light of this, the standard ICAO thrust setting of

<sup>3</sup> Time is expressed as a fraction of total roll time; fuel flow is expressed relative to fuel flow when the engine has stabilised at take-off thrust.

seven % was assumed during reduced-engine taxiing. It is worth noting that the PSDH made no specific recommendation for taking account of reduced-engine taxiing for NO<sub>x</sub> and PM emissions.

### Take-Off Thrust

3.3.45 The four thrust settings used in the ICAO databank were chosen to be representative of actual thrusts in the principal LTO flight phases, and early methodologies for calculating aircraft emissions simply assigned each LTO flight phase to one of the settings (with the exception of landing roll, where periods of reverse thrust were identified for some aircraft types), as shown in Table 3.3.1. However, more recent airport emission inventories recognise that large jets usually do not take off at 100 % thrust, with the actual thrust selected depending on take-off weight and air temperature. Typically, for large jets, actual take-off thrust lies between 75 % and 90 % of maximum thrust<sup>4</sup>.

**Table 3.3.1 : Thrust settings used in early emission inventories<sup>1</sup>**

Mode	Thrust
Taxi-out	7%
Holding at runway head	7%
Take-off roll	100%
Initial climb	100%
Climb-out	85%
Approach	30%
Landing roll <sup>2</sup>	7%
Taxi-in	7%

<sup>1</sup> These values have now been superseded by more detailed methodologies

<sup>2</sup> Periods of reverse thrust above idle were recognised even in early emission inventories

3.3.46 NO<sub>x</sub> emissions from take-off roll are a major component of the total ground-level NO<sub>x</sub> from aircraft at an airport, and the emission rate during take-off is strongly dependent on thrust, not only does fuel flow rate increase with thrust but the NO<sub>x</sub> emissions index (g NO<sub>x</sub> per kg fuel burned) also increases with thrust. Furthermore, there is large variability in the NO<sub>x</sub> emission indices from one engine type to another. Thus, it is important to make realistic estimates of the thrust settings for those

operator/aircraft type/engine combinations that have high utilisation at Gatwick Airport.

3.3.47 Actual take-off thrust settings are not routinely available on a flight-by-flight basis, although they can be extracted from FDR data. For PSDH, BA developed a methodology that enables information on take-off thrust to be derived from information on actual aircraft take-off weight. The methodology is based on their analysis of an extensive set of take-off thrust (derived from FDR data) and weight data for their fleet at Heathrow (Morris, 2002). BA found that, to a reasonable approximation, when flexible thrust<sup>5</sup> is being used the ratio of actual take-off thrust to maximum take-off thrust is given by the ratio of actual take-off weight (ATOW) to Performance Limited Take-Off Weight (PLTOW)<sup>6</sup>, subject to a lower limit set by regulation, normally 75 %.

3.3.48 Prior to the compilation of the 2002/3 Gatwick Airport emission inventory, British Airports Authority (BAA) carried out a survey of the principal airlines operating at Gatwick, first to ascertain how commonly flexible thrust (via the Assumed Temperature Method) was used at Gatwick and then to ask for information on ATOW and PLTOW for those operators using it. Airlines do not normally release ATOW on a flight-by-flight basis, but many of the major operators at Gatwick were willing to release annual-average ATOW information and were also willing to give information on the average limiting take-off weight. Due to a problem of terminology in the survey questionnaire, however, the airlines actually provided the lower of the PLTOW and the structural limit on weight (termed the Maximum Take-Off Weight, MTOW). Most aircraft types operating at Gatwick in typical weather conditions are not performance limited, so generally MTOW is less than PLTOW, so using the limiting weight values as provided tended to give a conservative (ie over-) estimate of mean take-off thrust.

3.3.49 Specific PLTOW<sub>0</sub><sup>7</sup> values were obtained for the fleets of BA and Air2000 operating at Gatwick. Two values were supplied for each aircraft type, corresponding to runway directions 26L and 08R, although the differences were typically less than two %. For other airlines, the potentially conservative nature of the estimates of mean thrust was accepted. It is not possible to use PLTOW<sub>0</sub> values from one airline for another for the same aircraft type because PLTOW<sub>0</sub> depends on details of the aircraft configuration,

in particular which engines are fitted. The indications from the BA and Air2000 data were that the degree of thrust overestimation would be generally less than five %.

3.3.50 Even if it is an airline's policy to use reduced thrust where possible, there are circumstances when 100 % thrust is mandated even if the aircraft is not at its limiting take-off weight, for example when the runway is icy or there is excessive low-level wind shear. Typically, the annual fraction of departures at 100 % thrust lies in the range of 2 - 10 %. Data on this fraction was requested in the BAA survey, and this fraction was treated separately in the emissions analysis.

3.3.51 In some instances, the airline indicated that for a given aircraft type a fixed thrust de-rate is used (sometimes called 'push-button de-rate'). In this case, the airline was asked to give the value of the de-rated thrust. De-rated thrust can be used in conjunction with the assumed temperature method, and if this was indicated in the survey response then the appropriate ATOW and limiting weight information was also requested.

3.3.52 Where survey results were not available for a given aircraft type<sup>8</sup> for a given airline, the value of mean take-off thrust was taken to be the average of the values obtained for the same aircraft type operated by other airlines (if possible with a similar type of business, ie scheduled or low-cost/charter). Small jets were assumed to take off at 100 % thrust.

3.3.53 For the 2005/6 inventory (Underwood *et al.*, 2008), major operators at Gatwick Airport were asked to update the information on average ATOW and PLTOW values for the principal aircraft types in their fleets operating at Gatwick in the relevant period (ensuring that the terminology problems of the earlier survey were not repeated) on the grounds that load factors and routes may have changed in the intervening period. BA, EasyJet, Ryanair and Great British (GB) Airways provided updated information for key aircraft types in their Gatwick fleets, and the corresponding thrust values were used for the inventory. The updated information covered around 50 % of the movements in the 2005/6 period. For other operators/aircraft types, the values used for the previous inventory were retained.

<sup>4</sup> All thrusts in the following text are expressed as a percentage of the rated output (F<sub>00</sub>), the maximum thrust available for take-off under normal operating conditions at ISA sea level static conditions.

<sup>5</sup> 'Flexible' thrust is a term used to contrast with push-button de-rated thrust and is typically applied via the 'Assumed Temperature Method'. In the latter, the aircraft flight management system is supplied with the value of the maximum air temperature at which the aircraft could

take off with its actual take-off weight, according to the flight manual. This is an approved method that maintains safety margins.

<sup>6</sup> PLTOW is the maximum take-off weight for a flight given by the aircraft flight manual, with due account taken of outside air temperature (OAT), wind speed/direction, runway characteristics (elevation, length, slope) and obstacle clearances. If it is higher than the maximum take-off

weight (MTOW) determined by structural considerations, then MTOW would set the limiting take-off weight for the flight.

<sup>7</sup> PLTOW<sub>0</sub> is the value of PLTOW for 15°C OAT and zero wind. This is used in the BA thrust methodology if actual average values of PLTOW are not available.

<sup>8</sup> 'Aircraft type' in this context refers to main type and series (ie B747-400); data for one series were not automatically assumed to apply to other series.

3.3.54	Ryanair indicated that it uses ‘push-button de-rate’ on its B737–800 aircraft (one of the two principal aircraft types operated by Ryanair during the period of interest, the other being the B737–200), and provided information that enabled the average amount of de-rate to be estimated. Ryanair also use flexible thrust on this aircraft, but previous Ryanair data indicated that this flexibility leads on average to little additional thrust reduction.		the climb-out thrust is set at take-off thrust; otherwise it was set at 85 %. It was recognised that this procedure was likely to overestimate climb-out NO <sub>x</sub> emissions, but emissions above the cut-back height have an insignificant influence on ground-level annual-mean concentrations (even when the potential influence of trailing vortices is taken into account), so the approximation was considered acceptable from a local air quality viewpoint.		updating of data sources were made for the subsequent inventories.
					<b>Approach</b>
3.3.55	For the current study, GAL undertook a survey of the airlines to update the information on take-off thrust. The responses to the survey contained sufficient information to update the assumptions for TUI, Thomas Cook, EasyJet and Virgin Atlantic. The take-off thrust assumptions for BA were retained from the 2005/6 inventory (Underwood <i>et al</i> , 2008).	3.3.59	However, the PSDH recognised that total emissions in the LTO cycle are also of interest beyond the local air quality perspective, for instance for the calculation of greenhouse gas emissions, and made recommendations aimed at improving estimates of elevated emissions, including recommendations on climb-out thrust, which are summarised below.	3.3.63	Data for the approach mode were obtained from Gatwick’s NTK system, which provides accurate positioning information every four seconds on a flight-by-flight basis. Sample NTK data, covering all arrivals for eight representative days from 2018, were used to derive average times in each phase of approach for a number of aircraft types. The sample data included both westerly and easterly operations from each season of the year. The data were available for the two approach segments (from 3000 ft to 2,000 ft and from 2,000 ft to the ground).
					<b>Landing Roll</b>
3.3.56	Where no specific data were available from any of the surveys for a particular aircraft type, the average value over all jet aircraft types with the same number of engines was used. This procedure for filling data gaps is consistent with that advocated by the PSDH.	3.3.60	Large commercial jets usually have several pre-set climb thrust settings, typically the maximum climb setting (CLB) and two lower settings, CLB1 and CLB2 (nominally 10 and 20 %, respectively lower thrust than CLB). The actual climb settings depend on aircraft type and engine fit, but for most types CLB does indeed appear to be close to 85 % of the full engine rating, with CLB1 and CLB2 at around 78 and 70 % of full rating. Thus, the PSDH report recommends the following procedure for setting climb-out thrust	3.3.64	For landing roll, GAL provided a sample of runway occupancy data from their ground radar system for August 2018. The data were flight-by-flight records including runway occupancy times (from threshold to runway exit to the nearest second) and an identification of the runway exit block. These times (and exit blocks) were matched to arrival records from GAL’s aircraft movement database. The runway occupancy data were also used to calculate average landing roll times by runway, exit block and aircraft type and to give exit block frequency by runway and aircraft type. These average times from the August 2018 sample were assigned to the remaining arrival records.
3.3.57	The above procedure gives thrust values based on annual average values of weight. In principle, PLTOW is influenced by ambient temperature, so that the take-off thrust for aircraft of a given take-off weight could show systematic diurnal and seasonal variations. However, modern commercial aircraft show little dependence of PLTOW on ambient temperature across the range of temperatures commonly experienced in the UK, so the influence of ambient temperature on take-off thrust for a given aircraft weight is not expected to be significant. Actual take-off weights for a given aircraft type operated by a given airline may also vary with time of day and season due to systematic variation in load factors or routes served, but the detailed ATOW data are not available to take this into account. The use of average weight data is unlikely to introduce significant error in the estimates of annual take-off emissions, but could influence the diurnal and seasonal profile of emissions.		<ul style="list-style-type: none"> <li>▪ use 85% for take-off thrust settings between 100 and 90%;</li> <li>▪ use 78% for take-off thrust settings between 90 and 80%;</li> <li>▪ use 70% for take-off thrust settings between 80 and 75% (the normal lower limit on take-off thrust); and</li> <li>▪ set climb-out thrust equal to take-off thrust if take-off thrust is less than 75% (for particular cases where an aircraft type is specifically certificated for take-off at less than 75%).</li> </ul>	3.3.65	<b>Reverse Thrust</b>
					From the airline survey undertaken for this assessment, estimates were obtained of the frequency and duration of reverse idle and reverse thrust above idle on an airline/aircraft fleet by fleet basis. These data have fed though to emission calculations, with suitable averaged data applied to those airline/aircraft fleets that detailed information was not available for.
					<b>Taxiing</b>
					3.3.66
3.3.58	In the standard ICAO LTO cycle, the thrust after cut-back is 85 %, but in practice aircraft use a range of thrust settings, with the value for a particular flight linked in part to the take-off thrust. In particular, the aircraft would not climb out at a thrust setting higher than at take-off. In the 2002/3 Gatwick Airport inventory, the influence of reduced-thrust take-off was recognised simply in terms of a constraint that if the take-off thrust is less than 85 %	3.3.61	These recommendations were adopted for the 2005/6 Gatwick inventory and have been retained for the 2018 inventory and future years.		Gatwick’s airport operational management system (IDAHO) provides, on a flight-by-flight basis, the times (to the nearest minute) of a number of key ‘events’; for example, for arrivals it gives the time the aircraft landed and the time it arrives at stand (On-Chox); for departing aircraft, it gives the time the aircraft left the stand (Off-Chox) and the time it became airborne. The IDAHO data align very closely with records from GAL’s aircraft
			<b>Times-in-Mode</b>		
		3.3.62	The PSDH report did not make any specific recommendations on how times-in-mode for the LTO flight phases should be assessed, but endorsed the AEA <sup>9</sup> approach of using ground-radar and Noise and Track-Keeping (NTK) data where available. An early version of this approach was used for the 2002/3 Gatwick inventory and further refinements to the methodology and the		

<sup>9</sup> AEA Technology was acquired by Ricardo Group, forming Ricardo-AEA Ltd, in 2012.

movement database, providing a match for 99.9 % of the movements.

3.3.67 Taxi-in times were calculated on a flight-by-flight basis, by subtracting landing-roll times from the total time from when the aircraft landed to the time it arrived at stand. Suitably averaged data were applied to the few unmatched records.

3.3.68 Taxi-out times were calculated on a flight-by-flight basis, by subtracting hold, line-up and take-off roll times from the total time from when the aircraft left the stand to the time it became airborne. Again, suitably averaged data were applied to the few unmatched records.

#### Hold, Line-Up and Take-Off Roll

3.3.69 In addition to providing runway occupancy times for arrivals, the sample runway occupancy data for August 2018 provided the runway holding time for departures, the time to line-up and the runway occupancy time (from lined-up to airborne). The data were provided to the nearest second and there was also an identification of the runway hold point (entry block). These times (and entry blocks) were matched to departure records from GAL's aircraft movement database. The data were used to calculate average holding, line-up and take-off roll times by runway, hold point and aircraft type and to give hold point frequency by runway and aircraft type. These average times from the August 2018 sample were assigned to the remaining departure records.

3.3.70 It is recognised that runway occupancy time may provide an over-estimate of take-off roll time, as there may be some delay at the runway head prior to the start of the take-off roll. However, the degree of over-estimation is considered to be negligible and does not affect the results of the assessment.

#### Initial Climb and Climb-Out

3.3.71 Data for the initial climb and climb-out modes were obtained from Gatwick's NTK system for a sample covering all departures for eight representative days in 2018. The data were used to derive average times in initial-climb and climb-out for a number of aircraft types and included both westerly and easterly operations from each season of the year.

3.3.72 It is understood that some operators/aircraft types normally cut back at 1,000 ft rather than 1,500 ft for noise-compliance reasons. Advice from the Civil Aviation Authority (CAA) during the PSDH work indicated that the lower cut-back was used by most aircraft in the 'Heavy' wake-vortex category (typically B777, B747, B767, A340, A310, A300, MD11) and by aircraft in the 'Medium'

wake-vortex category (typically B737, A319, A320, A321) for particular operators. The NTK data were further analysed to derive for times and distances to 1,000 ft for these aircraft types. All the remaining departures were assumed to cut back at 1,500 ft.

### 3.4 Aircraft Auxiliary Power Unit Emissions

3.4.1 APU emissions (kg) from a given aircraft movement were calculated as the product of the APU running time (s), the fuel consumption (kg per s) and the emission factor (kg pollutant per kg fuel consumed) appropriate to the APU model fitted on the aircraft.

3.4.2 There are relatively few openly-available sources of information giving APU emission factors (kg pollutant per kg fuel burned) and fuel flow rates (kg per hour), principally because APUs are not included in the ICAO certification process. The (United States) Federal Aviation Administration (FAA) reviewed the information available in 2000 by persuading the principal manufacturer (Honeywell) to comment on the datasets being recommended at the time by the FAA. The resulting set of APU emission indices, which have been widely employed in the compilation of airport emission inventories, were used for the 2002/3 Gatwick Airport inventory and quoted in the corresponding inventory report. No PM<sub>10</sub> emission factors were available, so a notional value of 0.1 g/kg fuel was used, based on the average value for main engines according to the methodology being used at the time.

3.4.3 Two limitations of the FAA data set were that (a) values are available for only a limited number of APU types that were common some years ago and that (b) the values given are averages for a typical APU cycle consisting of specified fractional amounts of various operational modes (such as providing electrical power only or providing air conditioning). This cycle (the details of which are not known for the FAA data) may differ from the actual cycle typical of Gatwick operations.

3.4.4 The release of detailed modal APU emission indices is controlled by the APU manufacturers, but data are released to aircraft operators for the purposes of generating emission inventories, provided the values for individual APU models are not published. For the work of the PSDH, a compromise was worked out whereby BA derived from the detailed manufacturer's data supplied to them a set of representative modal emission indices for general use in compiling inventories. This approach allowed greater realism to be reflected in the emission factors used for airport emission inventories whilst maintaining the level of

confidentiality required by the manufacturers. The key elements of this methodology have been adopted in the CAEP guidance report on airport emission inventories referred to earlier (CAEP, 2007).

3.4.5 Potentially there is a wide range of APU operating conditions for which differing fuel flow rates and emission factors apply, ranging from 'no load' through to the starting of main engines with the provision of electrical power to the aircraft systems. Other load conditions include the supply of electrical power and/or the provision of air conditioning. However, inspection of the data revealed that it is adequate to characterise APU operations in terms of three modes: (a) no load; (b) air conditioning plus electrical power (labelled ECS – environmental control systems - for convenience below) and (c) main engine start plus electrical power (labelled main engine start (MES) below).

3.4.6 For NO<sub>x</sub> emissions, BA defined six APU classes that adequately span the range of values found in the detailed data; each aircraft type was assigned to one of the six classes for the purpose of calculating APU NO<sub>x</sub> emissions. The modal NO<sub>x</sub> emission rates (product of fuel flow rate and emission index) for the six classes are given in Table 3.4.1 with the principal aircraft types assigned to the classes. It will be seen later that APU running times are dominated by the 'ECS' mode so overall emission indices are similar to those in this column of . As expected, these values span a similar range as the cycle-average values used in earlier inventories.

**Table 3.4.1: APU NO<sub>x</sub> Emission Rates and Class Assignments**

NO <sub>x</sub> Class	NO <sub>x</sub> Emission Rate (kg per hour)			Aircraft Types in Class
	NO Load	ECS	MES	
a	0.274	0.452	0.530	B727-100/200; BAe 146; A318; ERJ 135/145; F100, Tu 154M; Business Jets (with an APU)
b	0.364	0.805	1.016	B737-NG; CRJ; CRJ700; MD90
c	0.565	1.064	1.354	B737-CB757-2; A319/320/321; MD80; B767-2; B767-3
d	0.798	1.756	2.091	A300; A310; MD11; DC10; L1011-1/5/50/100

e	1.137	2.071	2.645	A330; B747-4; B747-SP; A340-3; B747-1; B747-2; B747-3
f	1.210	2.892	4.048	B777-2; B777-3; A340-6; A380

3.4.7 The detailed data on PM<sub>10</sub> emission indices proved more difficult to generalise, but BA found that the large variability in modal PM<sub>10</sub> emission rates could be reduced if the emission rates were expressed as a function of the corresponding NO<sub>x</sub> emission index. In this way, BA distinguished three classes of APU for which a different functional form of the relationship between PM<sub>10</sub> emission rate and NO<sub>x</sub> emission rate was appropriate, with each aircraft type assigned to one of these classes. The forms of the relationships derived are shown in Table 3.4.2 with the principal aircraft types assigned to the classes. PM<sub>2.5</sub> emission indices were set equal to the corresponding PM<sub>10</sub> indices.

**Table 3.4.2: APU PM<sub>10</sub> Emission Rates**

PM <sub>10</sub> Class	PM <sub>10</sub> Emission Rate (kg per hour) <sup>10</sup>	Aircraft Types in Class
A	PM <sub>10</sub> =0.0233 x (NO <sub>x</sub> )0.0934	All types (with an APU) except those below
B	PM <sub>10</sub> =0.379 x (NO <sub>x</sub> )2.642	Business jets (with an APU); BAe146; ERJ 135/145; CRJ; CRJ700
C	PM <sub>10</sub> =0.0630 x (NO <sub>x</sub> )0.173	B757-2; B767-2; B767-3; A300; A310

3.4.8 The Gatwick Airport Directive: GAD/F:28/17 sets limits on the use of aircraft auxiliary power units. It sets separate constraints on wide-bodied and narrow-bodied aircraft. For wide-bodied types, APU running time prior to scheduled departure time is limited normally to 50 minutes; running time on arrival at stand is limited normally to 15 minutes. For narrow-bodied jets, the equivalent times are 10 minutes on departure and 10 minutes on arrival.

3.4.9 In the absence of statistical data specific to Gatwick Airport, APU running times for previous inventories were also based on the equivalent limits in force at the time.

3.4.10 For this Project, logs of compliance audits undertaken during 2018 were made available. Other than for dispensations, these indicate that the limits set out in the directive are being observed. Furthermore, statistical analysis of the compliance logs and data on average turnaround times, suggest that, on-stand, APU are typically in operation for about 60 % of the limit times. There is a degree of uncertainty surrounding this percentage, however the derived times have been used in current and future modelling. The assumption is considered conservative for future cases as a result of likely pressure to reduce APU running times and implementation of initiatives including Fixed Electrical Ground Power (FEGP) and Pre-Conditioned Air Units (PCA)<sup>11</sup>.

3.4.11 The above procedure leads to total APU running time, whereas the PSDH methodology distinguishes three operating modes, namely (a) no load; (b) air conditioning plus electrical power (labelled ECS) and (c) main engine start plus electrical power (labelled MES), so the total time needs to be partitioned amongst these three modes. BA provided estimates of the typical times for the no-load and MES modes, with the former given as 180 seconds (all aircraft types) and the latter as 35 seconds for

two-engined aircraft or 140 seconds for four-engined aircraft. These times, which were applied to Heathrow in the PSDH work, have been adopted in the CAEP guidance report (CAEP, 2004), and were assumed to apply at Gatwick Airport. Thus, for arrivals, the time assigned to the ECS mode was set equal to the difference between total arrival running time and no-load time. For departures, the time assigned to the ECS mode was set equal to the time remaining after subtraction of no-load and MES times from the total departure running time.

3.4.12 With the increased use of reduced-engine taxiing there is a propensity for aircraft to operate their APUs during taxiing. In light of this, the survey that GAL undertook to identify the extent to which reduced-engine taxiing was used at the airport also asked about APU use during taxiing. The responses provided estimates of its frequency and duration for both arrivals and departures on an airline/aircraft fleet by fleet by fleet basis. These have fed though to emission calculations, with suitable averaged data applied to those airline/aircraft fleets that did not respond.

### 3.5 Engine Testing Emissions

3.5.1 An estimate of the emissions from engine testing on the airport was based on detailed logs of tests carried out during 2018 (total of 192 tests during the year). The logs provide information on the aircraft type, the aircraft registration number, the location of the test, the number of engines tested, an indication of the power setting of each engine tested and the total test duration. Emissions (g) for a given test were calculated as the product of a test time (s), the fuel flow rate of the relevant engine type (kg per s) at the appropriate thrust setting and the relevant emission factor (g pollutant per kg fuel consumed), summed over the engines involved in the test.

3.5.2 Power setting was specified using descriptive terms such as 'ground idle', 'flight idle', 'full power' etc., although for above-idle settings supplementary information was given on the actual thrust as a percentage of the engine rating (F<sub>00</sub> – the maximum thrust (engine rating) at sea-level in standard atmospheric conditions). The great majority of tests were at ground idle or flight idle. For ground idle, the PSDH-recommended reductions in fuel flow have been applied. The PSDH report also notes that 'flight idle' is typically 10 – 15 % F<sub>00</sub>, so a value of 15 % has been in this assessment.

<sup>10</sup> as function of NO<sub>x</sub> emission rate (kg per hour)

<sup>11</sup> Current practice demonstrates APU times at the airport are reduced through FEGP and PCA.

3.5.3 From discussions with aircraft operators, it is expected that engines are run at high power for periods of only a few minutes even if the total duration of the test period is much longer, but there is no information in the test logs on what fraction of the total run time was at high power. However, given that there were only 7 runs at above-idle power in the 2018 period, it was assumed conservatively that the whole run time was at the above-idle setting for these runs. A sensitivity test indicated that restricting the high-power running to five minutes per engine per test reduced the total NO<sub>x</sub> from engine testing by 15 %.

3.5.4 Engine type was assigned based on the aircraft registration number. For thrust settings intermediate between ICAO standard test thrust points (7 %, 30 %, 85 % and 100 %), the interpolation procedure described earlier in the context of reduced-thrust take-off was used. The PSDH factors for engine deterioration were also included.

### 3.6 Aircraft Brake and Tyre Wear

3.6.1 The 2002/3 Gatwick Airport emission inventory included an estimate of the contribution to PM<sub>10</sub> emissions from aircraft brake and tyre wear, albeit based on sparse data. The estimate was based on the generalisation of information obtained from a single operator at Stansted airport giving the amount of material eroded from brakes and tyres per landing for Fokker 100 and BAe146 aircraft. In the absence of any specific data, it was assumed that all eroded material would end up as suspended particulate matter in the PM<sub>10</sub> size range, recognising that this would almost certainly lead to an overestimation of PM<sub>10</sub> mass (given the blackening of runways and aircraft undercarriages). In order to estimate emissions from the whole fleet at Gatwick Airport based on this limited information, it was assumed that the PM<sub>10</sub> mass per landing would scale with the size of the aircraft, as represented by its MTOW, although there were no specific data to support this assumption.

3.6.2 More recently, additional information has become available to supplement the earlier data. Maintenance operators at Stansted airport have provided data on brake wear for the B737-300 and tyre wear for the B737 and A320, supplemented by data from the aircraft tyre manufacturers. Also, information on tyre wear has been compiled by BA for a number of the aircraft types in their fleet at Heathrow.

3.6.3 For the PSDH, QinetiQ reviewed the available data on brake and tyre wear and recommended a methodology for making best use of the information (Horton, 2006). For brake wear, the earlier

assumption that all the eroded mass ends up as suspended PM<sub>10</sub> particulate matter was retained, partly by analogy to road-vehicle data indicating that a significant fraction of the eroded mass can end up as PM<sub>10</sub>, but with continuing recognition that this is likely to lead to an overestimation of the PM<sub>10</sub> mass emitted. Similarly, the assumption that the emitted PM<sub>10</sub> mass per landing, scales with aircraft weight was retained. Pooling the data for the B737-300 with the earlier data gave an emission factor of  $2.5 \times 10^{-7}$  kg PM<sub>10</sub> per kg MTOW.

3.6.4 For tyre wear, the methodology was based principally on the BA information, which covered a wider range of aircraft size than previous data. This gave support to a linear dependence of mass eroded per landing on aircraft weight (represented as MTOW), and a linear regression of the data yielded the following relationship:

$$\begin{aligned} \text{Amount lost per landing (kg)} \\ &= 2.23 \times 10^{-6} \text{ MTOW} - 6 \times 10^{-6} \text{ (MTOW in kg)} \\ &- 0.0879 \text{ kg for MTOW greater than } 50,000 \text{ kg} \end{aligned}$$

3.6.5 The report gave no recommendation for modelling tyre wear for aircraft with MTOW less than 50,000 kg, and in implementing the above methodology in subsequent Heathrow emission inventories for the PSDH it was assumed that the eroded mass per landing varied linearly from the value at an MTOW of 50,000 kg given by the above to zero, at an MTOW of zero.

3.6.6 Judging by analogy to the road-vehicle data, QinetiQ considered it over-conservative to assume that all the eroded mass from tyre wear is suspended as particulate matter, and a PM<sub>10</sub> fraction of 10 % was assumed, which is at the upper end of the range observed for road-vehicle tyres. This contrasts with the assumption made for the 2002/3 Gatwick inventory that all eroded tyre material contributes to suspended PM<sub>10</sub> mass.

3.6.7 The above PSDH methodology (Department for Transport, 2006) was adopted for the 2005/6 Gatwick emission inventory and has again been used for the 2018 inventory. For 2018 the brake and tyre wear used actual arrival data. It is recognised that there remain significant uncertainties in estimating PM<sub>10</sub> emissions from brake and tyre wear, but these would only be reduced when more aircraft-specific data become available. The summed brake-wear and tyre-wear emission factor detailed above is around a factor of three, smaller than that used for the 2002/3 inventory, principally as a result of the less conservative assumption for the fraction of material suspended from tyre wear.

3.6.8 The mean size of particles from attrition processes such as brake and tyre wear tends to be much higher than from combustion processes, so in this case setting PM<sub>2.5</sub> emission factors equal to PM<sub>10</sub> emission factors is likely to significantly overestimate PM<sub>2.5</sub> emissions. There are no specific data on the PM<sub>2.5</sub>/PM<sub>10</sub> mass ratio for aircraft brake and tyres, so equivalent data for road vehicles were used, adding to the uncertainty in the PM<sub>2.5</sub> estimates. The road-vehicle values were taken from a review of brake and tyre wear carried out for the United Nations Economic Commission for Europe (UNECE, 2003). This estimates that the PM<sub>2.5</sub>/PM<sub>10</sub> mass ratio for brake wear is 0.4 and for tyre wear is 0.7; these ratios were adopted for aircraft brake and tyre wear for the 2005/6 Gatwick inventory (Underwood *et al.*, 2008) and have been retained for this current study.

## 3.7 Future Year Aircraft Emissions

### Movement Data

3.7.1 For each of the future case options, GAL provided fleet data in the form of annual forecasts of aircraft movements broken down by aircraft type and time of day (Day, Evening and Night). The diurnal profile of movements was derived from these forecasts by assuming a uniform distribution of movements within each period (Day, Evening and Night). In the absence of movement data for each day of the year, the annual profile of movements was assumed to be flat; this assumption is expected to be conservative as there are generally more flights in the summer months where dispersion conditions are more favourable. A summary of these forecasts is shown in Table 3.7.1.



Table 3.7.1: Annual Aircraft Movements

Aircraft	2029		2032		2038		2047	
	Without Project	With Project	Without Project	With Project	Without Project	With Project	Without Project	With Project
319	18,393	19,279	8,177	8,520	0	0	0	0
320	76,554	81,115	46,780	52,402	0	0	0	0
321	5,895	6,281	0	0	0	0	0	0
73H	11,080	11,814	1,582	1,661	0	0	0	0
AT7	489	489	0	0	0	0	0	0
320neo	91,661	96,852	125,718	153,076	178,618	211,073	162,952	191,040
321neo	20,878	22,238	32,570	39,308	36,072	42,794	52,392	62,905
738Max	33,347	35,523	42,738	47,083	44,288	48,165	44,629	48,274
737Max10	2,913	3,125	3,299	4,099	3,316	4,101	3,359	4,111
CS100	5,622	5,624	6,063	6,171	6,043	6,214	6,014	6,278
CS300	1,912	2,037	2,443	9,302	2,451	9,306	1,791	8,578
772	8,888	9,681	2,339	2,486	0	0	0	0
333	3,155	3,437	904	904	0	0	0	0
77W	235	256	0	0	0	0	0	0
788	5,074	5,504	6,201	9,201	6,521	9,199	7,446	9,866
789	17,959	19,560	26,772	34,026	31,896	39,575	36,284	42,634
359	4,203	4,578	5,208	7,703	5,716	8,092	6,695	8,802
350	1,450	1,580	806	806	1,827	1,871	2,091	2,045
77X	352	383	587	587	2,200	2,200	3,066	3,005
339neo	117	128	587	587	587	587	615	602
388	2,346	2,555	2,346	2,346	733	733	0	0
ER4	104	111	106	128	107	129	110	131
CJL	103	109	104	126	106	128	109	129
GS5	90	95	91	110	92	111	95	112
H28	62	66	63	76	64	77	66	78
CCJ	56	59	57	69	58	69	59	70
D2L	55	58	55	67	56	68	58	69
CJ1	45	47	45	55	46	55	47	56
HAP	60	63	60	73	61	74	63	75
EP3	35	37	35	43	36	43	37	44
Total	313,133	332,683	315,735	381,013	320,894	384,664	327,978	388,900

### Engine Assignment

- 3.7.2 The movement data provided by Gatwick Airport for movements in future years included generic aircraft types. The majority of the aircraft types included in the list are already in production; however, one type (the Boeing 777-X) is not yet in production and so there was a need to define the engine characteristics for this aircraft.
- 3.7.3 For existing aircraft types, the movement data for 2018 were used to define the percentage split between the different engine types. In the case of the Airbus A320neo family, which entered service in recent years and hence has only few movements in 2018, the split between engine manufacturers on the previous generation of the type (the A320ceo family) was taken to indicate the likely engine preferences for airlines as they transition their fleets to the new variants. Thus, the percentage split between CFM International (the CFM56-5B engines) and Pratt & Whitney (the V2500-A5 engines) on the A320ceo family aircraft were maintained when defining the split between the CFM LEAP-1A and PW1100G engines on the A320neo aircraft.
- 3.7.4 For the aircraft type not yet in production, the Boeing 777-X (specifically the -9X variant) the engine type for this aircraft has been announced (as the General Electric GE9X) but certification-based emissions data are not yet available (certification data are generally released for new engines once the aircraft and engine have entered service). Therefore, the emissions characteristics of the new engine were estimated using publicly-available data for this assessment:
- engine rated thrust: 470kN;
  - engine OPR: 60:1; and
  - engine specific fuel consumption (sfc): 10 % lower than the GE90-115B.
- 3.7.5 The engine was assumed to use a combustor based on the most advanced that General Electric (GE) currently has in production, that fitted to the GENx engine. Therefore, the fuel flow rates at the four certification test points were set to be 18 % lower than the equivalent values for the GE90-115B (combining 10 % lower sfc and 8.5 % lower rated thrust), while the emission indices for NO<sub>x</sub> were set to those for the GENx-1B76/P2.

- 3.7.6 The analyses described in paragraphs 3.7.2 to 3.7.5 defined the engine types (existing and future) which would be fitted to the aircraft operating at the airport in the future years and the proportions of the aircraft types fitted with the relevant engine types.
- ### Times in Mode
- 3.7.7 With the exception of taxiing and hold, the times-in-mode used for the 2018 baseline have been used for the future years. For reduced-engine taxiing and off-stand APU use the duration depended on the airline. The total duration for each aircraft type in 2018 was summed and pro-rated on the basis of the change in air transport movements (ATMs) for that aircraft type. The duration was then averaged across all aircraft for each aircraft type.
- 3.7.8 For the future cases, taxiing and hold times were obtained from airport simulation modelling. These gave times for westerly and easterly operations both with and without the northern runway in operation (cases with and without Project respectively). The taxi and hold times are shown in Table 3.7.2.

**Table 3.7.2: Taxi and Hold Times**

Mode	Time (minutes)			
	Without Project - Westerlies	Without Project - Easterlies	With Project - Westerlies	With Project - Easterlies
Taxi-in <sup>1</sup>	8.78	5.76	9.37 (8.97 in 2029)	6.19 (5.92 in 2029)
Taxi-out	12.04	13.53 <sup>2</sup>	9.58 (8.91 in 2029)	13.22 (12.84 in 2029)
Hold	7.15	7.15 <sup>2</sup>	6.37 (4.04 in 2029)	6.89 (4.86 in 2029)

<sup>1</sup> from touchdown

<sup>2</sup> only total taxi-out + hold modelled – hold time assumed same as for westerlies

- 3.7.9 The taxi-in times include the landing-roll times from touchdown to turn-off.

- 3.7.10 No airport simulation modelling was undertaken for easterly operations, so taxi and hold times were estimated from those for westerly operations. Hold times were assumed to be the same as for westerlies. Taxi-in and taxi-out times were assumed to be the same as taxi-out and taxi-in for westerlies, respectively.

### Take-off Thrust

- 3.7.11 Settings for reduced thrust on take-off are based on the Gatwick and BAA survey data that have been used to derive mean aircraft take-off thrust for each main aircraft type. The mean is calculated based on all movements for each aircraft type in the 2018 data.
- 3.7.12 New aircraft types were assigned suitable take-off thrusts based on averages of the 2018 data.

### Ambient Conditions

- 3.7.13 Corrections for ambient conditions, forward-speed effects and engine spool-up are based on PSDH but updated with new data.

### Runway Assignments

- 3.7.14 Runway assignments for a given hour of the year were the same as those used in the 2018 baseline in order to align with the meteorological conditions used in the dispersion modelling. The direction in which aircraft arrive and depart is largely determined by the wind direction, which of course also strongly affects the dispersion, so it is essential that the correlation between the two is preserved.
- 3.7.15 For two-runway options, movements also need to be assigned to the north or south runway. For all options with Project all arrivals are assigned to the southern runway. Departures are assigned to both runways, with all daytime departures of class C aircraft assigned to the northern runway and all other departures assigned to the southern runway.
- 3.7.16 The aircraft metrics for all of the main assessment and slow fleet transition (SFT) scenarios are presented in Table 3.7.3 and Table 3.7.4. The SFT scenario is based on an assumption that the rate of transition of Gatwick's airline fleet is slower than in 'the Northern Runway Project (NRP)' and 'Baseline' Cases with the same number of passenger and aircraft movements.

**Table 3.7.3: Aircraft Modelling Metrics – Main Assessment Scenarios**

Scenario	Average Of NO <sub>x</sub> Dp/Foo	M(NO <sub>x</sub> )	ER(NO <sub>x</sub> )100	ER(NO <sub>x</sub> )85	ER(NO <sub>x</sub> )30	ER(NO <sub>x</sub> )7
2018	43.0	13,169.8	76.8	48.6	7.3	1.1
2029 without Project	41.7	14,135.8	93.8	50.2	7.4	1.2
2029 with Project	41.8	14,265.0	94.7	50.8	7.5	1.2
2032 without Project	40.5	13,876.3	96.1	48.7	7.1	1.2
2032 with Project	40.3	13,964.2	96.4	49.0	7.1	1.2
2038 without Project	38.2	13,452.1	98.3	47.2	6.9	1.1
2038 with Project	38.3	13,590.0	98.3	47.6	6.9	1.2
2047 without Project	39.5	14,181.0	104.7	50.7	7.2	1.2
2047 with Project	39.4	14,150.2	103.4	50.4	7.1	1.2

**Table 3.7.4: Aircraft Modelling Metrics – Sensitivity Test Scenarios (Slow Fleet Transition)**

Scenario	Average Of NO <sub>x</sub> Dp/Foo	M(NO <sub>x</sub> )	ER(NO <sub>x</sub> )100	ER(NO <sub>x</sub> )85	ER(NO <sub>x</sub> )30	ER(NO <sub>x</sub> )7
2029 without Project SFT	42.4	14,157.6	89.9	51.0	7.6	1.2
2029 with Project SFT	42.5	14,286.8	90.8	51.5	7.6	1.2
2032 without Project SFT	41.9	14,315.9	92.8	51.2	7.5	1.2
2032 with Project SFT	41.8	14,383.9	93.2	51.4	7.5	1.2
2038 without Project SFT	41.2	14,403.1	101.4	51.2	7.3	1.2
2038 with Project SFT	40.9	14,398.6	100.6	51.1	7.3	1.2
2047 without Project	39.5	14,181.0	104.7	50.7	7.2	1.2
2047 with Project	39.4	14,150.2	103.4	50.4	7.1	1.2

### 3.9 Ground Support Equipment Emissions

- 3.9.1 This source category includes all vehicles and plant that generate exhaust emissions airside, principally vehicles associated with aircraft turn-around (vehicles operated by caterers, cleaners and fuel handlers, Ground Power Units and buses) but also vehicles associated with runway maintenance.
- 3.9.2 The energy team provided forecasts of fuel consumption for GAL and third-party vehicles. These included medium-ambition scenarios for the options with Project and baseline scenarios for the options without Project.
- 3.9.3 Emissions from ground support equipment were calculated from estimates of the annual amount of fuel used airside by various vehicle categories, with emission factors expressed as grams of pollutant per kg of fuel consumed.
- 3.9.4 For each of the future case options, the fuel consumption projections were used to scale activity data from 2018. The emission factors used for future year scenarios reflect the progression of the airside fleet with older vehicles being replaced by newer ones with tighter emissions standards.
- 3.9.5 The airport filling station, which supplies fuel to GAL, third party operators and staff, is the primary source of fuel used by vehicles operating airside, but it is also recognised that fuel obtained off-airport (for example brought in by caterers and cleaners with off-airport bases) is used airside. However, this additional source is assumed to be balanced out by GAL and third-party fuel obtained from the airport filling station that is used off-airport. All staff fuel is assumed to be used off-airport.

#### Fuel Apportionment

- 3.9.6 Each vehicle in the airside vehicle permit (AVP) database was assigned to one of eight principal categories, five for road vehicles (Articulated heavy goods vehicle (HGV), Car, Coach, light goods vehicle (LGV) and Rigid HGV) and three for off-road vehicles (37–75 kW, 75–130 kW and 130–560 kW), determined from information on the vehicle manufacturer and model. Every non-electric vehicle was assumed to have used an equal share, weighted by vehicle size, of the fuel dispensed by the airport filling station, with the proviso of petrol only being apportioned to light duty vehicles (Cars and LGVs).

### Emission Factors

#### Hot-Running Exhaust Emissions

- 3.9.7 Exhaust emission factors (g pollutant per kg fuel consumed) depend on vehicle category and the 'Euro' standard of the vehicle (ie the stage of EU emissions control to which the engine conforms). EU emission limits are different for road and off-road vehicles, both in terms of limit values and introduction dates.
- 3.9.8 Where possible, the Euro standard was derived from the vehicle registration number, assuming that the vehicle had the minimum Euro standard compatible with its year of registration. In practice, vehicles may be manufactured to a standard higher than the minimum and/or vehicles may be retrofitted with exhaust after-treatment that improves its emission performance over that at manufacture. On the whole, however, year of manufacture is an adequate indicator of Euro standard.
- 3.9.9 Where it was not possible to derive the year of registration from the vehicle registration number (commonplace for non-road vehicle categories) a weighted average emission factor was applied based on standards in place over the previous ten years (ie effectively assuming a uniform ten year age profile for each vehicle).
- 3.9.10 The emission-factor data set used takes account of Euro standards already included in EU Directives. For road vehicles, emission factors from COPERT 5 were used. The speed-emission curves include standards up to and including Euro 6 (Euro 6 is split into three stages: up to 2017, 2018–2020 and 2021+) for light duty vehicles (LDVs) and up to Euro VI for HDVs. Fuel consumption values and emission factors for NO<sub>x</sub> and PM<sub>10</sub> were worked out at 32 kph (corresponding to an airside speed limit of 20 mph); PM<sub>2.5</sub> emission factors were derived from the PM<sub>10</sub> emission factors using PM<sub>2.5</sub>/PM<sub>10</sub> ratios of 0.9 for catalyst-equipped petrol vehicles, 0.8 for non-catalyst petrol vehicles and 0.9 for diesel vehicles, as used in the National Atmospheric Emissions Inventory (NAEI) (Department for Business, Energy and Industrial Strategy (BEIS) and Defra, 2021).
- 3.9.11 For off-road (specialist) vehicles, exhaust emission factors for Uncontrolled, Stage I, Stage II, Stage IIIA, Stage IIIB and Stage IV diesel vehicles for NO<sub>x</sub> and PM (taken to be PM<sub>10</sub>) and PM<sub>2.5</sub> were taken from the latest issue of the EMEP/EEA Guidebook, available on the European Environment Agency website (EMEP/EEA, 2019). The values for Stages I to IV have been

taken from the emission limits in the EU Directive 2004/26/EC (European Commission, 2004).

#### Cold Starts

- 3.9.12 For NO<sub>x</sub> and PM<sub>10</sub>, the NAEI emission factor compilation contains data on 'cold starts' for LDVs, expressed as a quantity of pollutant per trip (BEIS and Defra, 2021). This represents the additional (integrated) amount of pollutant generated near the start of a trip, incurred during the period when the engine (and catalyst if fitted) has not yet reached its normal operating temperature range; this is particularly significant for catalyst-equipped vehicles. There are currently no cold start emission factors for HGVs.
- 3.9.13 It is difficult to estimate the number of cold starts associated with airside fuel use because of the wide range of duty cycles for airside vehicles and plant. However, even if every airside LDV had two cold starts every day, the contribution to annual NO<sub>x</sub> and PM emissions would be around 1 – 2 % of the total hot-running emissions. Thus, emissions from airside cold starts were disregarded.

#### Fugitive PM<sub>10</sub> and PM<sub>2.5</sub> Emissions

- 3.9.14 Four sources of fugitive PM<sub>10</sub> and PM<sub>2.5</sub> emissions from road vehicles have been included in the 2018 inventory: brake wear, tyre wear, road abrasion and re-suspended road dust. It is worth noting that fugitive emissions are becoming a significant component of total PM<sub>10</sub> and PM<sub>2.5</sub> emissions from road vehicles as exhaust emissions fall in response to tightening EU vehicle emission limits.
- 3.9.15 The fugitive-PM emission factors are expressed in terms of g per km and vary with vehicle category. For road vehicles operating airside, therefore, an estimate of the vehicle-km travelled for each vehicle was derived from the fuel consumed by the vehicle using the appropriate NAEI specific fuel consumption data at 32 kph. For off-road vehicles, it is expected that much of the fuel is used by stationary vehicles/plant so it is difficult to estimate corresponding fugitive-PM emissions. Rather than ignore the contribution, an upper bound on the contribution was included by converting all the fuel used into km travelled using the fuel consumption data for a road vehicle of comparable engine size. This is likely to overestimate the PM emissions from these vehicles by a significant factor, but in practice the resulting emissions contribution is not dominant.

### Heating Plant Emissions

- 3.9.16 Emissions from a given heating plant (g per year) were calculated as the product of the total amount of fuel used, expressed as the energy equivalent of the fuel in MJ per year, and an emission factor (g per MJ).
- 3.9.17 GAL supplied the annual fuel consumption (in kW-hr) for their facilities for 2018. All the boilers run on natural gas and the facilities listed span a wide range of annual consumptions, with only the North Terminal Boiler House and South Terminal Boiler House having an annual consumption of more than 10<sup>7</sup> kW-hr. GAL also supplied annual fuel consumption (kW-hr) for the Hilton Hotel and estimates were made for other airport facilities including hotels and hangars.
- 3.9.18 No NO<sub>x</sub> or PM stack emission measurement data were available for any of these boilers, so default emission factors (g per MJ) for NO<sub>x</sub> and PM<sub>10</sub> were taken from the EEA Guidebook (EMEP/EEA, 2019). Separate emission factors are given for various categories of fuel usage: for natural gas burning in boilers, the category 'other industrial combustion – natural gas' was selected.
- 3.9.19 The energy team provided forecasts of natural gas consumption for GAL and third parties and, separately, for standalone third parties. These included medium-ambition scenarios for the future year scenarios with and without the Project. For each of the future year scenarios, the natural gas consumption projections were used to scale emissions from 2018.
- 3.9.20 Additionally, GAL supplied the total food tonnage processed by their energy from waste facility in 2018. For the future year with Project scenarios, the energy from waste plant would be relocated. The location of the source has been updated for the dispersion modelling using data provided by GAL.
- 3.9.21 Emission factors (g per MJ) for NO<sub>x</sub> and PM<sub>10</sub> were derived from stack emissions monitoring undertaken in 2017 by Environmental Scientifics.

### Fire Training Ground Emissions

- 3.9.22 The Fire Training Ground (FTG) is included here for the sake of completeness, although the annual emissions of the pollutants of interest are expected to be negligible compared to those from other airport sources, based on previous emission inventories.

GAL provided the information that 44,404 litres of liquefied petroleum gas (LPG) was used in fire training activities during 2018.

- 3.9.23 LPG is usually a mixture of butane and propane predominantly, in varying proportions depending on the origin, but the emission factor data available are not detailed enough to vary with composition. There are no emission factors specific to the type of operation at the FTG, but it was judged that the NO<sub>x</sub> and PM<sub>10</sub><sup>12</sup> emission factors from AP-42 (United States Environment Protection Agency, 1995) for the burning of LPG in commercial boilers (0.1 to 3.0 MW) would be reasonably appropriate.

- 3.9.24 There are no specific data on the PM<sub>2.5</sub>/PM<sub>10</sub> ratio for open burning of these fuels, and it was conservatively assumed that the PM<sub>2.5</sub> mass is equal to the PM<sub>10</sub> mass. Given the extremely small PM<sub>10</sub> contribution from the FTG, this approximation has an insignificant impact on the estimate of the total airport-related PM<sub>2.5</sub> emissions.

- 3.9.25 Future year emissions from the fire training ground have been kept the same as in 2018 as it is an activity that is independent of the number of ATMs or passengers.

## 3.10 Road Traffic Emissions

### Highway Network

- 3.10.1 Traffic data was provided by the transport consultants. Annual average daily traffic (AADT) and traffic data representing the average conditions occurring in four specific time periods (morning peak, inter-peak, afternoon peak and off-peak) – traffic period data were provided (Table 3.10.1).
- 3.10.2 The traffic data takes into account embedded design mitigation in the Surface Access Strategy designed to reduce vehicle numbers. The air quality assessment uses the total vehicle numbers but does not consider further any detailed breakdown of fleet type associated with the airport and has used the EFT to provide the fleet mix for the modelled roads. Therefore, a conservative approach has been taken as the airport aspires to encourage and facilitate a shift to electric vehicles and go beyond the mode share targets where this is possible.
- 3.10.3 For the time periods vehicle speeds in kilometres per hour (kph) were provided. As two AM peak period speeds were provided,

the lowest of these were used to represent the speed for the AM traffic period. In absence of an off-peak period speed, the inter-peak period speed was used to represent the off peak traffic period.

**Table 3.10.1: Traffic time periods**

Traffic period	Time period
AM peak (AM)	3 hours (07.00 – 10.00)
Inter-peak (IP)	6 hours (10.00 – 16.00)
PM peak (PM)	3 hours (16.00 – 19.00)
Off peak (OP)	12 hours (19.00 – 07.00)

**Table 3.10.2: Speed time periods**

Speed period	Time period
AM peak (AM1)	1 hour (07.00 – 08.00)
AM peak (AM2)	1 hour (08.00 – 09.00)
Inter-peak (IP)	7 hours (09.00 – 16.00)
PM peak (PM)	2 hours (16.00 – 18.00)

- 3.10.4 The data comprised a fleet mix of cars, LGVs and HGVs split between airport related and non-airport related traffic. Airport-related traffic includes passenger cars, LGVs and HGVs related to the airport's operations, buses, coaches and staff cars.
- 3.10.5 Road traffic emissions for NO<sub>x</sub>, PM<sub>10</sub> and PM<sub>2.5</sub> were calculated using the factors from the Defra Emissions Factors Toolkit (EFT) version 11 (Defra, 2021) for the assessment of base, construction and operational traffic scenarios.
- 3.10.6 The EFT provides the latest predictions for vehicle emission rates up to 2050. The most recent analysis, at time of writing, indicates that "assumptions built into the EFT suggest that, on balance, the EFT is unlikely to over-state the rate at which NO<sub>x</sub> emissions decline in the future at an 'average' site in the UK. In practice, the balance of evidence suggests that NO<sub>x</sub> concentrations are most likely to decline more quickly in the future, on average, than predicted by the EFT." (Air Quality Consultants, 2020a).

<sup>12</sup>The AP-42 value applies to 'filterable particulate matter', which is assumed to be all PM<sub>10</sub>.

- 3.10.7 EFT v11 contains basic vehicle split composition data up to 2050<sup>13</sup> and this would be used to reflect the ongoing improvements in emissions from vehicles, in line with the government's commitment to transition to zero emission cars and vans (HM Government, 2020) and the banning of petrol and diesel vehicle sales in 2035. The latest version of the EFT has also resolved the previous concerns around underestimated emissions. This is acknowledged by Air Quality Consultants Ltd, developers of the Calculator Using Realistic Emissions for Diesels (CURED) model, who state it is no longer needed as the Defra EFT is now considered representative of actual emissions (Air Quality Consultants, 2020b).
- 3.10.8 The EFT provides the latest predictions for vehicle emission rates and vehicle split composition data up to 2050 for England (not London) roads and up to 2030 for London roads. Therefore, for the 2032, 2038 and 2047 assessment years, the corresponding emissions data were used for England (non-London) roads and for London roads 2030 emissions were used. The data post 2030 was provided by Defra to help support carbon assessment and it is important to note that uncertainty in the emission factors exists in the data from 2030 to 2050. However, with the known policy requirements to reduce emission from transport such as the Transport Decarbonisation Plan (Department for Transport, 2021) being in place in that period freezing all emissions at 2030 is considered overly conservative. This approach was discussed and agreed with stakeholders as noted in **ES Appendix 13.3.1: Summary of Stakeholder Scoping Responses** (Doc Ref. 5.3).
- 3.10.9 The pNO<sub>2</sub> emissions were derived from NO<sub>x</sub> using the percentages stated in NAEI for the relevant model year, as presented in Table 3.10.3 (BEIS and Defra, 2021).

**Table 3.10.3: Fraction of NO<sub>x</sub> Emitted by Vehicles as pNO<sub>2</sub>**

Year	pNO <sub>2</sub> Fraction
2018	0.286
2024	0.272
2029	0.240
2030	0.234

- 3.10.10 Emissions were calculated separately for each vehicle class and then added together for each road link split into airport and non-airport related traffic.
- 3.10.11 Speed data in kilometres per hour were provided for all traffic links from the transport consultants. Speeds at junctions and roundabouts where greater than 20 kph were modelled at a reduced speed (20 kph) to reflect queuing and congestion, in accordance with the Defra LAQM Technical Guidance (TG22) guidance (Defra, 2022).
- 3.10.12 National Highways have developed a tool to account for the additional contribution of ammonia (NH<sub>3</sub>) emissions from vehicles to deposited nitrogen (National Highways, 2022). This has been used to determine the nitrogen deposition at designated ecological sites assessed.
- 3.10.13 Assumptions and limitations with regards to the road traffic data are discussed in Section 7. The traffic data were the outputs of the Simulation and Assignment of Traffic to Urban Road Networks (SATURN) model, which used manual and automatic data count points as input. The geometry of the road network for the baseline, construction and operational traffic scenarios is presented in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.1.
- Car Parks**
- 3.10.14 Information on car park movements was provided by the transport team in the form of daily number of vehicles (cars) entering and leaving each car park for the existing and future year scenarios. Assumptions and limitations of this data are presented in Section 7. Emissions were calculated following the Cambridge Environmental Research Consultants (CERC) note on modelling car parks for both street-level and multi-storey car parks (CERC, 2017).
- 3.10.15 Emission factors for vehicles were taken from the latest Defra EFT, while cold start emissions were taken from the NAEI database (BEIS and Defra, 2021). The percentage of pNO<sub>2</sub> emissions was also taken from the NAEI and is presented in Table 3.10.3 (BEIS and Defra, 2021). A speed of 5 kph was assumed in all car parks.

- 3.10.16 The location of car parks included in the assessment for the baseline and future year scenarios are presented in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.2 to Figure 4.1.6.

### 3.11 Central Area Recycling Enclosure (CARE) facility

- 3.11.1 The design of the CARE facility has been assessed based on the details in **ES Chapter 5: Project description** (Doc Ref. 5.1). Some assumptions were required for the purpose of this assessment which are set out in this section. Therefore, whilst the exact details of the CARE facility are not confirmed at the time of preparing this report, the likely worst-case scenario for the proposed CARE facility is a doubling of capacity compared to the existing CARE facility (with one biomass boiler). As such, the emissions from two biomass boilers have been included, which are based on the specification of the existing biomass boiler.

#### Stack Height Assessment

- 3.11.2 A stack height assessment has been undertaken to determine a suitable height for the proposed biomass boiler stack. The stack height assessment is completed based on the data provided by project team and a stack emissions monitoring report (ESG, 2017).
- 3.11.3 The stack height assessment was completed using a cartesian grid of 5km by 5km at ground level with a resolution of 25m. The model grid extent is 524217, 138920 to 529217, 143920.

#### Emission Parameters

- 3.11.4 The biomass boilers parameters have been provided by the project team and emission data have been taken from a stack emissions monitoring report (ESG, 2017) which contains emission data for the currently operating CARE facility. They are presented in Section 4. Assumptions and limitations of this data are presented in Section 7. The process contributions (PC) from the CARE facility are based on a 24/7 operation and the same PCs have been applied to all relevant assessment years.
- 3.11.5 The parameters for the two units of biomass boiler as part of the CARE facility are presented in Table 3.11.1 and location is shown

<sup>13</sup> The EFT contains the factors for England (not London), London data does not include future factors beyond 2030.

in Figure 4.1.27. The stack height used in the assessment was determined by carrying out a stack height assessment, as noted.

**Table 3.11.1: CARE facility parameter and emissions**

Parameter	Unit	Biomass boilers
Stack location	NGR (m)	526714.4, 141414.8
Number of biomass boiler	-	2
Stack flue diameter	m	0.47
Stack height	m	48
Flue gas velocity	m/s	15
Flue exit temperature	°C	63
Moisture content	%	7.6
Oxygen content	%	7.6
Operation profile	-	24/7
NO <sub>x</sub>	g/s	0.5
SO <sub>2</sub>	g/s	0.08
PM <sub>10</sub>	g/s	0.004
PM <sub>2.5</sub>	g/s	0.004
CO	g/s	0.3
VOC	g/s	0.009

#### Plume Visibility assessment

- 3.11.6 Water in the emitted gases can condense in the air and form a visible plume if conditions are suitable. There are no formal or informal standards for visible plume lengths although visible plumes that are long enough to reach ground level should be avoided. A plume visibility assessment has been carried out using the ADMS 5.2 dispersion model. The frequency of visible plumes has been predicted.
- 3.11.7 Plume visibility from the stack depends on ambient meteorological conditions, flue gas humidity and the efflux temperature of the stack. Condensation of water droplets occurs when the temperature of the ambient air mixed with the flue gas, is lower than the saturation temperature of that mixture. If enough condensation occurs then a plume may become visible.
- 3.11.8 The water mass used in the modelling was 0.04 kg/kg.

#### Sensitivity Analysis

- 3.11.9 In order to define the method used to undertake the assessment a number of sensitivity analyses have been undertaken to determine which modelling options should or should not be included in the main assessment. Parameters and emissions from the CARE facility were used and the effect of changing elements of the modelling methodology were examined. The following has been considered as part of the sensitivity analysis:
- selection of met year from Gatwick Airport meteorological station (5 years examined); and
  - consideration of buildings.
- 3.11.10 The impact on ground level concentrations for a range of pollutants and averaging periods was examined using the maximum predicted on the grid of receptors.
- 3.11.11 The results are presented in **ES Appendix 13.9.1: Air Quality Results Tables and Figures – P3** (Doc Ref. 5.3).
- ### 3.12 NRMM and concrete batching plants
- 3.12.1 Non-Road Mobile Machinery (NRMM) equipment has been assessed based on details in **ES Chapter 5: Project description** (Doc Ref. 5.1) and **ES Chapter 14: Noise and Vibration Appendix 14.9.1: Construction Noise** (Doc Ref. 5.3). The Construction Noise Appendix provides quantities of different types of NRMM based on construction design information as discussed in **ES Chapter 14: Noise and Vibration** (Doc Ref. 5.1). The NRMM vehicles were grouped together to give a quantity for each general type of NRMM. The power output for each type of NRMM were assumed based on typical equipment on the market and from previous experience. Table 3.12.1 provides a summary of the NRMM usage. Each and every NRMM were assumed to be in operation 24 hours a day, 7 days a week and were all included for both 2024 and 2029 construction scenarios as a worse case assumption. In reality, each and every NRMM are very unlikely to be operating 100% of the time. The NRMM were assumed to meet Euro Stage V emission standards, which is considered a realistic assumption for future construction. The emission rates were based on Euro Stage V standards and the power output assumed for the type of NRMM. The percentage pNO<sub>2</sub> emissions were assumed to be the same as for a Euro 5 diesel car (16%), taken from the NAEI.

- 3.12.2 The Construction Noise Appendix indicated that five large concrete batching plants and one small batching plant would be considered. To provide a conservative assessment, it was assumed that the batching plants would be in operation 24 hours a day, 7 days a week. The concrete batching plants were assumed to be powered by a 500kW diesel generator based on a typical setup using a diesel engine rather than powered by alternative fuels or power taken from a grid connection. The technical specification of Cummins 500kW diesel engine (QSX15-G8) was used as an assumption for the generator powering the concrete batching plant, as a generic generator for that size. The emission for the Cummins diesel engine were used to estimate emissions. In addition to the batching plants, four generators were included to align with the assessment in the Noise NRMM assessment (**ES Appendix 14.9.1: Construction Noise** (Doc Ref. 5.3)), they would likely be used for other equipment and welfare facilities and were also assumed to be the same specification as the batching plant generators.
- 3.12.3 The assessment modelled each of the five work areas (Surface Access Construction, North Terminal, South Terminal, Northwest Airfield Construction and Southeast Airfield Construction) as area sources in ADMS. The work areas are shown in Figure 4.1.28. The assumptions set out mean the same emission rate has been applied across each area, which is considered to be a reasonable assumption as most plant will be transient within the areas and the exact locations for concrete batching plant has not yet been defined. All equipment in Table 3.12.1 is assumed to be operational 24/7, hence the model represents a pessimistic scenario.

**Table 3.12.1: Summary of NRMM equipment predicted**

NRMM Vehicle Type	Quantity	Fuel Type	Power Output (kW)
Excavator	37	Diesel	235
Rigid HGV	22	Diesel	239
Bulldozer	7	Diesel	126
Roller	12	Diesel	95
Tipper Truck	13	Diesel	350
Mobile Crane	14	Diesel	370
Concrete Mixer Truck	15	Diesel	125
Piling Machine	6	Diesel	400

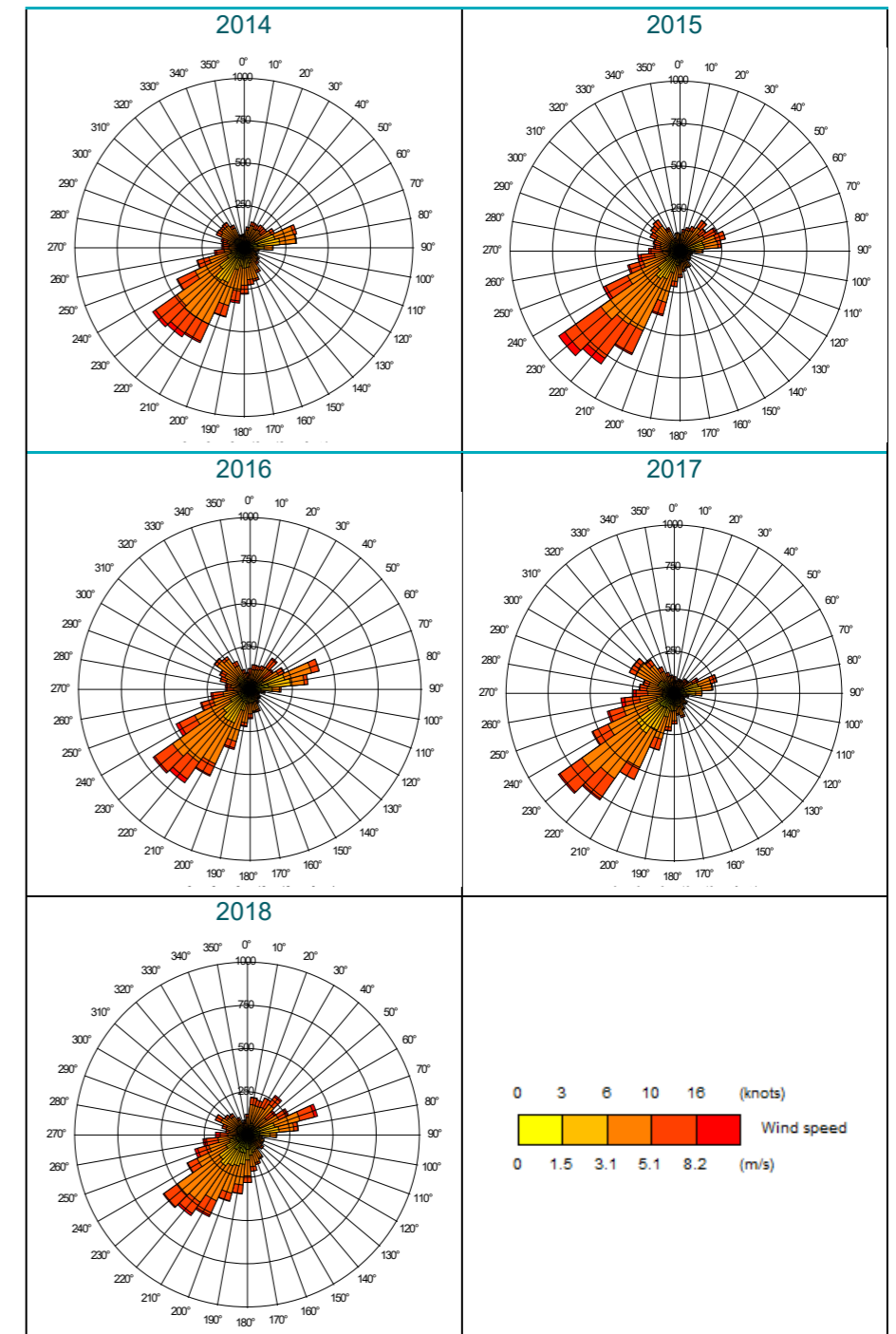
NRMM Vehicle Type	Quantity	Fuel Type	Power Output (kW)
Dumper 8T	3	Diesel	55
Dumper 40T	10	Diesel	350
Paving Machine	6	Diesel	140
Compressors/pumps	6	Diesel	129
Generator	4	Diesel	500
Batching Plant (Small)	1	Diesel	500
Batching Plant (Large)	5	Diesel	500

4.1.4

model. This is important when considering predictions of high percentiles and the number of exceedances. The guidance recommends that meteorological data should only be used if the percentage of usable hours is greater than 75 % and preferably greater than 90 %.

The 2014 – 2018 meteorological data from Gatwick Airport includes over 90 % of usable data (2014: 92 %; 2015: 94 %; 2016: 92 %; 2017: 96 % and 2018: 98 %). This is above the 90 % threshold and these data therefore meet the requirement of the Defra guidance. Diagram 4.1.1: Windrose for Gatwick Airport presents the windrose for the 2018 meteorological data from Gatwick Airport. It can be observed that prevailing winds are south westerly.

Diagram 4.1.1: Windrose for Gatwick Airport (2014 – 2018)



## 4 Model Setup

### 4.1 Model Setup Parameters

4.1.1 The Atmospheric Dispersion Modelling System (ADMS) ADMS-Airport (version 5.0.0.1) and ADMS 5.2 (CERC, 2016 and 2020) were used for this assessment. The ADMS software is widely used for air quality assessments in the UK and ADMS-Airport was the software used for the previous assessments at Gatwick airport.

#### Meteorology

4.1.2 The air quality dispersion model uses hourly sequential meteorological data from which to calculate the boundary layer parameters. Meteorological data from Gatwick Airport were obtained for 2018 for use in this assessment. For the CARE facility, five years of meteorological data were used. This is considered to be industry best practice for the assessment of point source emissions.

4.1.3 Most dispersion models do not use meteorological data if they relate to calm wind conditions, as dispersion of air pollutants is more difficult to calculate in these circumstances. The ADMS-Airport model treats calm wind conditions by setting the minimum wind speed to 0.75 m/s. Defra's LAQM (TG22) guidance (Defra, 2022) states that the meteorological data file is tested by running the meteorological pre-processor of the dispersion model and the relevant output log checked to confirm the number of missing hours and calm hours that cannot be used by the dispersion



### Other Model Parameters

4.1.5 The extent of mechanical turbulence (and hence, mixing) in the atmosphere is affected by the surface/ground over which the air is passing. Typical surface roughness values range from 0.0001 metres (for water or sandy deserts) to 1.5 metres (for cities, forests and industrial areas). In this assessment, a surface roughness of 0.2 metres was used for the meteorological and dispersion sites.

4.1.6 Another model parameter is the Monin-Obukhov length, which describes the minimum level of turbulence in the atmosphere, which can be limited due to the urban heat island effect. For this model, a minimum length of 20 metres was used.

### Buildings

4.1.7 Buildings can have a significant effect on the dispersion of pollutants in the vicinity of the CARE facility and have been included within the model. Building input geometries are shown in Table 4.1.1: and Figure 4.1.26 including nearby significant buildings. Buildings included in the assessment have been determined by a sensitivity test, and the worst-case scenario used. The main building has also been determined through the sensitivity test. The complex building geometry has been simplified so as to be included within the model which only accepts rectangular or circular building shapes.

**Table 4.1.1: Building geometries**

Building name	X	Y	Height (m)	Length (m)	Width (m)	Angle (°)
CARE building 1	526674	141411	15	20.3	30.5	166.6
CARE building 2 (main)	526746	141427	15	18.0	68.0	166.0
Hangar 9	526957	141301	11	62.5	93.4	166.6
Cargo building	526883	141366	10	584.5	46.0	76.1
Pier 7	526537	141208	18	20.4	677.4	165.2
Aircraft hangar	526113	141218	32	137.3	88.6	166.0
ASIG building	526855	141647	10	64.4	15.7	166.2

Building name	X	Y	Height (m)	Length (m)	Width (m)	Angle (°)
Fuel farm building	526961	141578	8	40.7	22.8	164.8
Fuel tanks 1	527041	141778	13	27.7	67.2	170.3
Fuel tanks 2	527025	141731	13	37.2	122.1	166.8

## 4.2 Spatial Representation

4.2.1 For some sources, for example taxiing, the emissions occur at well-defined spatial locations, in this example along taxiways. For other sources, such as airside vehicles, the location of the emissions is less well defined. For such sources, the total emissions have been calculated and have then been disaggregated over the area within which they typically occur using a surrogate parameter; for airside vehicles, the parameter is the product of aircraft movements and MTOW (see paragraph 4.2.19).

### Aircraft-Related Emissions

#### Aircraft Jet Sources

4.2.2 For modelling purposes aircraft were grouped into modelling categories (MCATs) of aircraft-engine combinations with similar dispersion characteristics, primarily geometry and plume buoyancy. A lead aircraft and representative engine was selected for each aircraft category. The MCATs, lead aircraft and representative engines are presented in Table 4.2.1.

**Table 4.2.1: Aircraft modelling categories**

MCAT	Typical Aircraft Type	Representative Engine
0	Piston and turboprop aircraft	N/A <sup>1</sup>
1	A319	CFM56-5B5/P
2	A320	CFM56-5B4/3
3	A321	CFM56-5B3/P
4	B757-200	RB211-535E4
5	B787-900	Trent 1000-J2
6	B777-200	GE90-85B
7	B777-200	Trent 895
8	B747-400	CF6-80C2B1F

MCAT	Typical Aircraft Type	Representative Engine
9	A380-800	GP7270
10	A320 neo	LEAP-1A26/26E1

<sup>1</sup> Piston and turboprop aircraft were modelled as passive releases (ie no jet buoyancy characteristic).

### Taxiing and Hold

4.2.3 The taxiway system on the airport was represented by a network of nodes joined by straight-line links. Each taxiing route was composed of a series of straight-line segments.

4.2.4 For the purpose of modelling taxiing routes, taxi-out from all stands in a given stand group to a given hold point were represented by a single taxiing route, taken from a representative point within the stand group. Taxi-out emissions assigned to a given taxi-out route were then distributed uniformly along the route.

4.2.5 A similar approach was used for taxi-in emissions. Taxi-in routes were devised for each runway exit/stand group pair and are shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figures 4.1.7 to 4.1.9.

4.2.6 Similarly, holding emissions for a given hold point were assigned to a line source joining the taxiway to where aircraft would join the runway for the corresponding hold point (Figure 4.1.10 to 4.1.12 in **ES Appendix 13.4.1: Air Quality Assessment Methodology**).

4.2.7 Due to evolving design decisions during the project the final location of rapid exit taxiways and taxiway Juliet are different to those modelled. As noted in the **ES Chapter 5: Project Description** (Doc Ref. 5.1) for the first exit in either runway direction the actual footprints would be located within an area 100m either side of the indicative position shown on Figure 5.2.1a to enable flexibility. The change would not alter the conclusions of the air quality assessment as the emissions have been captured and the distance to receptors is enough to allow for sufficient dispersion of emissions.

### Take-Off Roll and Landing Roll

4.2.8 Take-off roll emissions for a given flight were distributed along the runway between a start-of-roll point and a wheels-off point (Figure 4.1.13 in **ES Appendix 13.4.1: Air Quality Assessment Methodology**). As a result of engine spool-up and the forward-

speed effect, the acceleration of the aircraft is not constant; this has been taken into account in the model using the data provided in the ADMS model '.sec' file which spatially distributes the roll emissions.

- 4.2.9 Landing-roll emissions were distributed between the touchdown point and runway exit (Figure 4.1.14 and Figure 4.1.15 in **ES Appendix 13.4.1: Air Quality Assessment Methodology**), assuming a constant deceleration from a touchdown speed of 130 knots to a taxiing speed of 15 knots.

#### Initial Climb, Climb-Out and Approach

- 4.2.10 Climb profiles were stylised as two straight-line segments: from the end of roll to throttle-back (at 1,000 ft or 1,500 ft) and from throttle-back to 3,000 ft. Departure tracks were assumed to continue in the direction of the runway up to 3,000 ft (Figure 4.1.14 in **ES Appendix 13.4.1: Air Quality Assessment Methodology**). Aircraft may start to turn below this height, but the positional deviation caused by the approximation would only affect emission contributions that have an insignificant impact on ground-level concentrations.
- 4.2.11 The NTK data were analysed to give the average distances to reach throttle-back height and to reach 3,000 ft for each aircraft type. These were used to work out a mean initial climb angle and a mean climb-out angle for each aircraft group.
- 4.2.12 Approach emissions were represented as two co-linear line segments aligned with the runway (from 3,000 ft height to 2,000 ft height and then from 2,000 ft height to touch down) at a 3° angle to the horizontal. The total emissions for each segment were distributed uniformly along the corresponding line segment.

#### Brake and Tyre Wear

- 4.2.13 Brake and tyre wear during landing were represented in the model as volume sources on the runway. Tyre wear emissions were modelled at the point of touchdown only with actual data used for 2018 baseline. Appropriate assumptions were used for the future years with the northern runway as a dedicated departure runway. The modelled brake and tyre wear locations are shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology**, Figure 4.1.16 and 4.1.17.

#### APU Emissions

- 4.2.14 On-stand APU emissions were calculated separately for each stand as GAL's aircraft movement database included flight-by-

flight data on-stand used (including stands in the maintenance area). A volume source (50 metres × 50 metres × 12 metres) was located at each stand.

- 4.2.15 Off-stand APU emissions were assigned to the devised taxi-in and taxi-out routes.
- 4.2.16 The locations of the modelled APU emissions are shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.18 to 4.1.20.

#### Engine Testing

- 4.2.17 Engine testing is not a significant source of emissions compared with other on-site sources. The test log used for calculating emissions from engine ground runs gave the location of individual tests, identified as particular named or numbered areas on the airport. At the time of the production of the PEIR EGR locations for the Project had not been confirmed, and Engine testing was modelled at a maintenance area in the north of the airport to provide for a conservative assessment. The modelled locations are as shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.21. A volume source (50 metres × 50 metres × 15 metres) was modelled at each location.

- 4.2.18 Since production of the PEIR, EGR locations have been confirmed, however, the modelled locations have been kept as per those used in the PEIR. The Alpha 2 EGR site, has not been included in the current model (however four test locations are still modelled), as noted in **ES Chapter 5: Project Description** (Doc Ref. 5.1). It is only expected to be used 5% of the time and the change would not alter the conclusions of the air quality assessment. It is also important to note that no engine testing would be carried out at the maintenance area location used in the modelling, as all EGRs will take place on the taxiway system to the south of this.

#### Airside Support Vehicles/Plant

##### Ground support equipment

- 4.2.19 Airside vehicle emissions were assigned to stands in proportion to the 'airside activity' at the stands. To calculate airside activity, each aircraft movement was assigned a 'weight' to represent its contribution to airside activity in terms of demand for airside services. The weighting factor was taken to be the MTOW for the aircraft. Emissions associated with a stand were assigned to a volume source (50 metres × 50 metres × 3 metres) at the stand.

The locations of the modelled ground support equipment are shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.22 and 4.1.24.

#### Heating Plant

- 4.2.20 Emissions from the boiler houses and the energy from waste plant were treated as point sources. The boiler houses (one at the North Terminal and one at the South Terminal) are shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.25 and 4.1.26.

#### Fire Training Ground

- 4.2.21 Emissions from the fire training ground were assigned to a volume source (50 metres × 50 metres × 20 metres) located as shown on Figure 4.1.21 in **ES Appendix 13.4.1: Air Quality Assessment Methodology**.

#### Road Traffic

##### Highway Network

- 4.2.22 Emissions from road traffic are modelled as road sources. The ArcGIS geospatial software was used to assist in inputting road link information into the air quality model. The modelled roads are shown in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.1.
- 4.2.23 Road widths for each road link were calculated in ArcGIS geospatial software by snapping the road link layer to the OS Mastermap topographic layer. Larger interchange junctions and slip roads were treated individually to ensure the outermost extent of the carriageways were used. Sensitivity checks were carried out on all road links, to show outliers and these were manually investigated and adjusted using satellite imagery if required.
- 4.2.24 The road link features were checked for snapping in the ArcGIS geospatial software, to ensure all road features connected to each other without any gaps. The number of roads connected to a single point was calculated by producing end points for each road link. Any points with only one link associated were updated for snapping or checked to be the end of a road link.

##### Car Parks

- 4.2.25 Emissions from street level car parks were modelled as area sources and emissions from multi-storey car parks were modelled

as volume sources. The location of the modelled car parks are presented in **ES Appendix 13.4.1: Air Quality Assessment Methodology** Figure 4.1.2 to 4.1.6.

### 4.3 Temporal Variation

4.3.1 Temporal variation refers to variations during a day (diurnal variation) and/or between seasons. The temporal variation of emissions is represented in the dispersion model by use of temporal profiles. The level of detail needed in temporal profiles depends on the significance given to peak short-period concentrations and how these are estimated, which are matters to be considered at the dispersion modelling stage. Annual-mean concentrations are less sensitive to the details of the temporal profiles.

4.3.2 The highest resolution of temporal variation (shortest time period) that can be modelled in ADMS-Airport is the time resolution of the meteorological data, which is one hour.

#### Aircraft-Related Emissions

4.3.3 Aircraft exhaust emissions in the LTO flight phases were calculated at a time resolution of one hour based on the hourly data supplied in the 2018 baseline. This variation automatically incorporates diurnal and seasonal changes in the number and type of aircraft movements, systematic variations in ground-movement times-in-mode and the impact of diurnal and seasonal variations in ambient temperature and pressure. In the modelling of future years, the temporal variation was simplified as described in Section 4.3.

#### Airside Support Vehicles/Plant

4.3.4 Airside vehicles emissions were distributed among stands in proportion to the 'airside activity' (product of movements and aircraft MTOW), which is derived from the breakdown of aircraft movements by stand. These data were also used to provide temporal profiles of airside-vehicle emissions that vary with stand.

4.3.5 Other sources, such as the boiler-house emissions and the fire training ground, were assigned a uniform temporal profile.

4.3.6 No temporal variation was applied to car parks as the data were unavailable for this assessment.

### 4.4 Results Processing

4.4.1 Model verification was used to compare modelled pollutant data with measured real-world concentrations to assess the performance of the model and determine adjustment factors where required in accordance with Defra guidance (Defra, 2022). The model verification results are detailed in **ES Appendix 13.6.1: Air Quality Data and Model Verification** (Doc Ref. 5.3).

#### NO<sub>x</sub> to NO<sub>2</sub> Conversion

##### Road sources

4.4.2 The model predicts roadside NO<sub>x</sub> concentrations and therefore a suitable NO<sub>x</sub> to NO<sub>2</sub> conversion needs to be applied to the modelled concentrations. The method used for this conversion in the assessment follows the approach described by Clapp and Jenkin (Clapp and Jenkin, 2001), which takes account of the proportion of pNO<sub>2</sub> in the balance between NO and NO<sub>2</sub> and derives total NO<sub>2</sub> concentrations as a function of distance from major sources.

4.4.3 The method requires a value for the regional background oxidant, which was taken to be 33.5 parts per billion (ppb) in 2008 (Clapp and Jenkin, 2001) and was projected to increase by +0.1 ppb/year for future years, giving a value of 34.5 ppb for 2018, 35.1 ppb for 2024, 35.6 ppb for 2029, 35.9 for 2032, 36.5 for 2038 and 37.4 for 2047.

4.4.4 Defra provide a NO<sub>x</sub> to NO<sub>2</sub> calculator which is used primarily for the conversion of modelled road NO<sub>x</sub> emission to NO<sub>2</sub>. The Clapp and Jenkin approach allows different percentages of pNO<sub>2</sub> to be assumed depending on the source type, which is considered appropriate for this assessment which includes many other sources than just roads. However a sensitivity test was undertaken using the Defra calculator for comparison and the results are provided in **ES Appendix 13.9.2: Air Quality Sensitivity Tests** (Doc Ref. 5.3).

#### CARE Facility

4.4.5 The air quality model predicts concentrations of NO<sub>x</sub> which is a mixture of NO<sub>2</sub> and NO. Both gases react in the atmosphere, particularly with ozone. In general, the NO<sub>x</sub> are mainly emitted as NO and this converts to NO<sub>2</sub> in the atmosphere. The air quality standard has been set for NO<sub>2</sub> and therefore it is important that an appropriate conversion rate is used to calculate ambient NO<sub>2</sub> concentrations at the receptors that result from the modelled NO<sub>x</sub> emissions. It is proposed that the EA advice<sup>14</sup> on conversion rates is used, which suggests a ratio of 35% for short-term (ie hourly average) and 70% for long-term (ie annual mean) concentrations. In practice, these ratios represent conditions some distance away from a release source. Close to an industrial source, the proportion of NO<sub>2</sub> in NO<sub>x</sub> is typically much lower than this. Applying these ratios would therefore provide a conservative assessment.

#### Background concentrations

4.4.6 The Defra website (Defra, 2021) includes estimated background air pollution concentrations for each 1 km by 1 km OS grid square in the UK up to the year 2030. The background concentrations for the modelled receptors assumed that concentrations were frozen at 2030 and are presented in **ES Appendix 13.6.1: Air Quality Data and Model Verification** (Doc Ref. 5.3). This presents a conservative assumption for future years as backgrounds would be expected to reduce with technology improvements and take-up of electric vehicles.

4.4.7 It should be noted that twice the background concentration has been used for calculating the total short-term concentrations associated with the CARE facility, in line with Defra guidance.

4.4.8 For the ecological assessments, the background nitrogen deposition has been decreased by 1.12% per annum. This is taken from data presented for England in Table 4.2 of Annex 4 of the JNCC Report 665 Nitrogen Futures (JNCC, 2020), taking a 2017 baseline compared to a 2030 BAU scenario (ie with no additional mitigation to nitrogen deposition beyond those already part of policy). The use of this figure has been agreed with Natural England (NE) in its response to the consultation on the Preliminary Environmental Information Report (PEIR) where it confirms that it is appropriate.

<sup>14</sup> Environment Agency – Air Quality Modelling and Assessment Unit: Conversion ratios for NO<sub>x</sub> and NO<sub>2</sub>

## 5 Odour Impact Assessment

- 5.1.1 Odour is a mix of volatile chemical compounds (or a single compound) that triggers a reaction in the nose. As the nose is very sensitive it often only requires very low concentrations to trigger this reaction. Any odour, whether considered to be pleasant or unpleasant, can result in a loss of amenity for occupiers of property if it is unwanted. However, as noted in the Defra Odour Guidance for Local Authorities (Defra, 2010) when exposed to odour that are perceived to be unwanted these cause occupants of the area to have a “negative appraisal” of their environment. They cope with this stress in several ways, for instance, by changing behaviour, complaining, or seeking distractions from the odour source.
- 5.1.2 Several factors determine whether an odour is perceived by an individual as unpleasant, the Defra guidance notes the following as important:
- offensiveness of the odour;
  - intensity of the odour;
  - duration of exposure;
  - frequency of exposure; and
  - tolerance and expectation of the exposed subjects.
- 5.1.3 Odour concentrations are reported as European Odour Units per cubic metre (ouE/m<sup>3</sup>). One ouE/m<sup>3</sup> is the concentration at which 50% of an odour sampling panel can detect the odour. To measure the odour concentration, a sample is presented to an “odour panel” at various dilutions until only 50% of the panel can detect the odour. If the odour sample has had to be diluted by a factor of 10 then the original sample is considered to have an odour concentration of 10 ouE/m<sup>3</sup>.
- 5.1.4 There is no relevant guidance for assessment of odours in an internal environment, however, the IAQM guidance on Odours (IAQM, 2018) does recommend that where detailed modelling is not possible a semi-quantitative assessment is carried out using different assessment methods such as using the Source, Pathway, Receptor (SPR) model. As such this assessment uses a SPR approach and review of complaints data to inform the assessment along with feedback received during consultation.

### Source Pathway Receptor Assessment

- 5.1.5 The SPR approach examines each of the three factors for each potential odour source and receptor and then determines the risk of adverse odour impacts. This approach is largely for planning purposes where a new odorous process is proposed near to sensitive receptors (or vice versa). It is considered relevant to this assessment to review the potential change in odour as a result of the Project.
- 5.1.6 The IAQM guidance (IAQM, 2018) suggests that the following factors are considered for the SPR as shown in Table 5.1.1. This approach has been used for the assessment of baseline and future operations at the airport.

**Table 5.1.1: Risk factors for SPR approach**

Source Odour Potential	Pathway Effectiveness	Receptor
<p>Factors affecting the source odour potential include:</p> <ul style="list-style-type: none"> <li>▪ The magnitude of the odour release;</li> <li>▪ How inherently odorous the materials are; and</li> <li>▪ The unpleasantness (or offensiveness) of the odour.</li> </ul>	<p>Factors affecting the odour flux to the receptor are:</p> <ul style="list-style-type: none"> <li>▪ Distance from source to receptor;</li> <li>▪ Frequency of winds from the source to receptor (not relevant for internal odour sources);</li> <li>▪ The effectiveness of any mitigation/control to reduce the odour flux to the receptor; and</li> <li>▪ Topography and terrain.</li> </ul>	<p>Some receptors are more sensitive. This is largely determined by the expectations for the area.</p>

- 5.1.7 Typically, the greatest potential for adverse odour to occur is during periods of stable atmospheric conditions with calm or low wind speeds, generally when wind speeds are less than 3m/s. This reduces dilution and mixing of odours with ambient air and results in higher odour concentrations at receptor locations.

- 5.1.8 There are no prescribed distance criteria in relation to odour emissions. Therefore, the following distance ranges, based on distance from the potential distances to receptors have been used to define the effectiveness of the pathway:
- Receptors within 200 metres of the source;
  - Receptors 200 metres – 500 metres from the source; and
  - Receptors 500 metres – 1 kilometres from the source.
- 5.1.9 The percentage that the wind is blowing from the airport towards the receptor, with a speed of less than 3m/s, has been calculated. A 45° range of wind directions centred on the identified receptor has been used to ensure that a spatial extent of the airport was captured and takes into account the uncertainty of the measured wind directions and the plume width from the source.
- 5.1.10 This calculation used five years of meteorological data from Gatwick Airport (Diagram 4.1.1). From this calculation and the distance between the source and nearest identified receptor, the pathway effectiveness has been calculated.
- 5.1.11 Table 5.1.2 and Table 5.1.3 present the matrices extracted from the IAQM guidance (IAQM, 2018), which show the interaction between the source potential, odour pathway and sensitivity of receptors to derive the magnitude of risk of odour exposure. This has been used to determine the significance of any odour affects sensitive receptors.
- Qualitative Odour Assessment**
- 5.1.12 The initial step in the assessment is to estimate the odour generating potential of the activity, considering the magnitude of the release, how inherently odorous it is, and the relative unpleasantness of the emission. The “*pathway effectiveness*” is then determined, by considering the distance from the source, the frequency of exposure considering prevailing winds where appropriate, the likely effectiveness of dispersion and terrain between the emission point and receptor location.

**Table 5.1.2: Risk of odour exposure at a specific receptor location**

Pathway Effectiveness	Source Odour Potential		
	Small	Medium	Large
Highly Effective	Low Risk	Medium Risk	High Risk
Moderately Effective	Negligible Risk	Low Risk	Medium Risk

Ineffective	Negligible Risk	Negligible Risk	Low Risk
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5.1.13 Finally, a judgement on the significance of the effect on receptors is then made. The matrix in describes the general relationship between the risk of odour exposure (impact) experienced by a receptor for a given sensitivity and the magnitude of adverse effect that is likely to result.

**Table 5.1.3: Likely Magnitude of Odour Effect at a Specific Receptor Location**

Risk of Odour Exposure	Receptor Sensitivity		
	Low	Medium	High
High	Slight adverse	Moderate adverse	Substantial adverse
Medium	Negligible	Slight adverse	Moderate adverse
Low	Negligible	Negligible	Slight adverse
Negligible	Negligible	Negligible	Negligible

5.1.14 If the overall effect is described as moderate or substantial, the effect is considered to be significant.

5.1.15 This does not mean that the potential odour activities are unacceptable, rather it is an indication that careful consideration should be given to the consequences of the emissions and the scope for mitigation measures that should be brought forward.

5.1.16 Where the overall effect is judged to be slight adverse or negligible, this wouldn't be considered to be significant, in line with industry guidance (IAQM, 2017).

## 6 Health Impact Assessment Methodology

6.1.1 An interpolated receptor concentration for the air pollutants of NO<sub>2</sub>, PM<sub>10</sub> and PM<sub>2.5</sub> in each operational scenario has been calculated for use in the health assessment reported in **ES Chapter 18: Health and Wellbeing** (Doc Ref. 5.1). The air quality emissions and modelling methodology for airport-related sources and road traffic using AADT data detailed in this document has been used for the air quality modelling for the health assessment.

6.1.2 AddressBase Plus data (Ordnance Survey, 2022) were obtained from Ordnance Survey (OS) to identify all residential property

locations across the wider study area. The air quality modelling was undertaken at a 100m resolution grid across the wider study area and pollutant concentrations at residential property locations were interpolated from the modelled grid.

6.1.3 The modelled concentration at each residential property was multiplied by the average population at each residential property, which was assumed to be 2.6 and is consistent with the population statistics used by the noise consultants for the project. The change in the interpolated receptor concentration used for the health assessment describes the difference between the With and Without Project scenarios for the assessment year.

6.1.4 The interpolated receptor concentration results were considered in the health assessment in **ES Chapter 18: Health and Wellbeing** (Doc Ref. 5.1). The details of the health assessment methodology, with regards to the air quality related effects, is detailed in **ES Chapter 18: Health and Wellbeing** (Doc Ref. 5.1).

## 7 Assumptions and Limitations

**Table 7.1.1: Assumptions and Limitations of the Air Quality Assessment**

Project Item	Assumption/Limitation
Road traffic	For road links outside of London, vehicle emissions for 'England (not London)' have been used in the Defra EFT tool. For road links in London, vehicle emissions for 'London' have been used in the Defra EFT tool.
Car parks	No temporal profile has been applied to car parks. Cold start emissions have been calculated, assuming all cars are diesel, providing a conservative estimate. The BA car park is assumed to have the same number of movements as Car Park W for the 2018 Baseline scenario. It is assumed that the number of movements are the same, assuming no growth in future years. This is due to the similar size and category of the car park, as advised by the Project transport consultants. Staff Car Park J was operating as Car Rental in 2018 and Do Minimum Operational scenarios, so was removed for these scenarios. Car Park H is assumed to be a multi storey car park from the Do Something 2029 scenario, where the car park doubles in capacity. Car Park Y is completed as a multi storey car park in 2038 scenarios. The multi storey extent is assumed to be the same as the surface level, four storeys were assumed for the multi storey car park based on a four times increase in spaces. Each storey of a car park, if decking or multi-storey, is assumed to be 3 m. Vehicle speed is assumed to be 5 kph for all car parks. The higher number of daily in/out movements provided by the Project transport consultants was used for calculations of emissions to provide a conservative estimate for all car parks.

Project Item	Assumption/Limitation
	If the in/out movements provided by the Project transport consultants were for road links that may be entry/exit points for multiple car parks the number of movements were distributed across the car parks according to gross floor area. Car parks with decking were modelled as a multi-storey car park, represented in the ADMS model as a volume source.
Construction dust	Trackout has been assessed for access to contractor compounds and any routes taken by HGVs assuming all entrances are used. Measures to mitigate trackout are included in the Code of Construction Practice (CoCP).
NRMM	A conservative approach has been taken regarding construction phase mitigation, for example all NRMM has been assessed as being Euro 5 diesel standards, however as noted in Chapter 13 the Project commits to using low or zero emissions equipment where possible.
Heating plant emissions	It is assumed that the heating plant emissions would be dominated by those servicing the needs of on-airport buildings therefore only heating plants that are sited within the current airport perimeter are included in the airport inventory.
Airport construction sources	Any construction dust sources of PM <sub>10</sub> or PM <sub>2.5</sub> on the airport during the period of interest are not included in the airport emissions inventory, these are addressed in a separate construction dust assessment. Emissions from NRMM are included.
Taxi-out emissions	The assessment assumes all engines are lit during pushback, due to lack of specific information on when engines are lit for each aircraft type and operator. It is assumed that all engines are shut down immediately when the aircraft reaches the stand. It is judged that each assumption would compensate the other.
Aircraft engine type	If there was no engine type identifier available a default engine based on the most common engine for that aircraft type was used. If there was no

Project Item	Assumption/Limitation
	data providing an engine for a particular aircraft type, a typical engine according to standard aircraft reference sources was assigned to the aircraft.
Aircraft emission factors for PM <sub>10</sub>	The ICAO databank contains measured non-volatile PM <sub>10</sub> emission factors for only a small number of newer engines. For older engines, the methodology in CAEP guidance was used to derive non-volatile PM <sub>10</sub> emissions. The guidance was also used to estimate volatile sulphate and organic PM <sub>10</sub> emissions for all aircraft engines.
Aircraft PM <sub>2.5</sub> exhaust emissions	It was assumed that the mass of PM <sub>2.5</sub> in aircraft exhaust equals the mass of PM <sub>10</sub> (for both volatile and non-volatile components).
Aircraft emissions of pNO <sub>2</sub>	Aircraft emissions of pNO <sub>2</sub> were derived from the fractions presented in the PSDH methodology. These factors were 4.5 % pNO <sub>2</sub> at 100 % thrust, 5.3 % at 85 % thrust, 15 % at 30 % thrust and 37.5 % at seven % thrust. Linear interpolation was used for intermediate thrust settings.
Aircraft Engine spool-up	NO <sub>x</sub> emission index for all engines and aircraft types was kept constant during the transient phase as that applicable at take-off thrust so the net effect of spool-up on estimated emission rate derives solely from the lower fuel flow rate. The effects of engine spool-up has been ignored for PM <sub>10</sub> and PM <sub>2.5</sub> in line with the PSDH recommendation.
Aircraft taxiing	Fuel flow rates for engine types other than Rolls Royce were estimated to be set 17.5 % lower, and for Rolls Royce engines 32.5 % lower than the ICAO seven % value because survey results suggested lower thrust settings were used. These values applied to all periods of taxiing and hold. The NO <sub>x</sub> and PM <sub>10</sub> emission indices at the lower fuel flow rate were held the same as the value at seven % thrust.
Aircraft take-off thrust	Take-off thrusts for BA used the 2005/6 inventory. Updated survey was undertaken for TUI, Thomas Cook, EasyJet and Virgin Atlantic data with

Project Item	Assumption/Limitation
	aircraft using the average value over all jet aircraft types with the same number of engines was used.
Aircraft climb-out thrusts	The following thrusts were used in this assessment: 85 % for take-off thrust settings between 100 % and 90 %; 78 % for take-off thrust settings between 90 % and 80 %; 70 % for take-off thrust settings between 80 % and 75 % (the normal lower limit on take-off thrust) and set climb-out thrust equal to take-off thrust if take-off thrust is less than 75 % (for particular cases where an aircraft type is specifically certificated for take-off at less than 75 %).
Aircraft initial climb and climb-out	Sample NTK data from Gatwick, covering all departures for eight representative days from 2018, were used to derive average times in initial-climb and climb-out for a number of aircraft types. For defined 'Heavy' and 'Medium' aircraft types, the NTK data were analysed for times and distances to 1,000 ft rather than 1,500 ft.
Aircraft brake and tyre wear	Brake and tyre wear was calculated using methodology from the Gatwick 2005/6 emissions inventory and used the same PM <sub>2.5</sub> fractions of PM <sub>10</sub> (40 % for brake wear and 70 % of tyre wear).
GSE	All staff fuel is assumed to be used off-airport. The Euro standard was derived from vehicle registration number, assuming that the vehicle had the minimum Euro standard compatible with its year of registration. Where registrations were not available a uniform ten-year age profile for each vehicle was assumed.
FTG	It was conservatively assumed that the PM <sub>2.5</sub> mass is equal to the PM <sub>10</sub> mass for open burning of LPG.
Aircraft take-off roll and landing roll	Landing-roll emissions were distributed between the touchdown point and runway exit, assuming a constant deceleration from a touchdown speed of 130 knots to a taxiing speed of 15 knots.
Aircraft departure tracks	Departure tracks were assumed to continue in the direction of the runway up to 3,000 ft.

Project Item	Assumption/Limitation
Future aircraft diurnal profiles	The diurnal profile of movements was assumed using a uniform distribution of movements within each period (Day, Evening and Night). In the absence of movement data for each day of the year, the annual profile of movements was assumed to be flat as a conservative assumption.
CARE facility	The building information used in the dispersion model is from both the Project team and Google Earth. It is considered to be the best available data. The emission rate for PM <sub>2.5</sub> is assumed to be the same as PM <sub>10</sub> . The flue stack of the current CARE facility is fitted with a cone in order to increase the exit velocity to 15m/s, it is assumed that a cone would also be fitted in the proposed CARE facility and the exit velocity would also be 15m/s.

## 8 References

### 8.1 Legislation

Directive 2004/ 26/EC of the European Parliament and of the Council of 21 April 2004 amending Directive 97/68/EC on the approximation of the laws of the Member States relating to measures against the emission of gaseous and particulate pollutants from internal combustion engines to be installed in non-road mobile machinery.

The Air Quality Standards (Amendment) Regulations 2016, SI2016/1184.

The Air Quality Standards Regulations 2010, SI 2010/1001.

### 8.2 Published Documents

Air Quality Consultants (2020a) Performance of Defra's Emission Factor Toolkit 2013 – 2019 [online source]. Available at: <https://www.aqconsultants.co.uk/CMSPages/GetFile.aspx?guid=7fba769d-f1df-49c4-a2e7-f3dd6f316ec1>

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## 9 Glossary

### 9.1 Glossary of Terms

**Table 9.1.1 Glossary**

Term	Description
AADT	Annual Average Daily Traffic
ADMS	Atmospheric Dispersion Modelling System
AEROCERT	Aircraft Environmental Impacts and Certification Criteria
APU	Auxiliary Power Unit
ATM	Air Transport Movements
ATOW	Actual Take-off Weight
AVP	Airside Vehicle Permit
BA	British Airways
BAA	British Airports Authority
BEIS	Business, Energy and Industrial Strategy
CAA	Civil Aviation Authority
CAEP	Committee on Aviation Environmental Protection
CARE	Central Area Recycling Enclosure
CERC	Cambridge Environmental Research Consultants
CLB	Climb setting
CO	Carbon monoxide
CO <sub>2</sub>	Carbon dioxide
CoCP	Code of Construction Practice
CURED	Calculator Using Realistic Emissions for Diesels
Defra	Department for Environment, Food and Rural Affairs
DCO	Development Consent Order – planning consent process for Nationally Significant Infrastructure Projects
DfT	Department for Transport
EASA	European Union Aviation Safety Agency
ECS	Environmental Control Systems
EEA	European Environment Agency
EFT	Emissions Factors Toolkit
EIA	Environmental Impact Assessment

Term	Description
EMEP	European Monitoring and Evaluation programme
ES	Environmental Statement
FAA	Federal Aviation Administration
FDR	Flight Data Recorder
FEGP	Fixed Electrical Ground Power
FOCA	Federal Office of Civil Aviation
FOI	Swedish Defence Research Agency
FTG	Fire Training Ground
GAL	Gatwick Airport Limited – the company which operates Gatwick Airport
GB	Great British
GE	General Electric
GIS	Geographic Information System
GSE	Ground Support Equipment
HC	Hydrocarbons
HDV	Heavy Duty Vehicles
HGV	Heavy Goods Vehicle
HRA	Habitat Regulations Assessment
IAQM	Institute of Air Quality Assessment
ICAO	International Civil Aviation Organisation
IDAHO	Gatwick’s airport operational management system
ISA	International Standard Atmosphere
LDV	Light Duty Vehicle
LGV	Light Goods Vehicle
LPG	Liquefied petroleum gas
LTO	Landing and Take-off
MCATs	Modelling categories
MES	Main Engine Start
MTOW	Maximum Take-Off Weight
NAEI	National Atmospheric Emissions Inventory
NO	Nitric oxide
NO <sub>2</sub>	Nitrogen dioxide
NOAA	National Oceanic and Atmospheric Administration
NO <sub>x</sub>	Oxides of nitrogen
NRMM	Non-Road Mobile Machinery
NTK	Noise and Track-Keeping



Term	Description
O <sub>3</sub>	Ozone
OAT	Outside air temperature
Off-chox	The time an aircraft leaves a stand
On-chox	The time an aircraft arrives at a stand
OPR	Overall Pressure Ratio
PAHs	Polycyclic Aromatic Hydrocarbons
PCA	Pre-Conditioned Air Units
PEIR	Preliminary Environmental Information Report
PLTOW	Performance Limited Take-Off Weight
PM <sub>10</sub>	Airborne particles that have a median diameter of 10 microns
PM <sub>2.5</sub>	Airborne particles that have a median diameter of 2.5 microns
pNO <sub>2</sub>	Primary nitrogen dioxide
ppb	Parts per billion
PSDH	Project for the Sustainable Development of Heathrow
SATURN	Simulation and Assignment of Traffic to Urban Road Networks
sfc	Specific fuel consumption
SN	Smoke number
SO <sub>2</sub>	Sulphur Dioxide
UID	Unique Engine Identifier
UK	United Kingdom
UNECE	United Nations Economic Commission for Europe
VOCs	Volatile Organic Compounds